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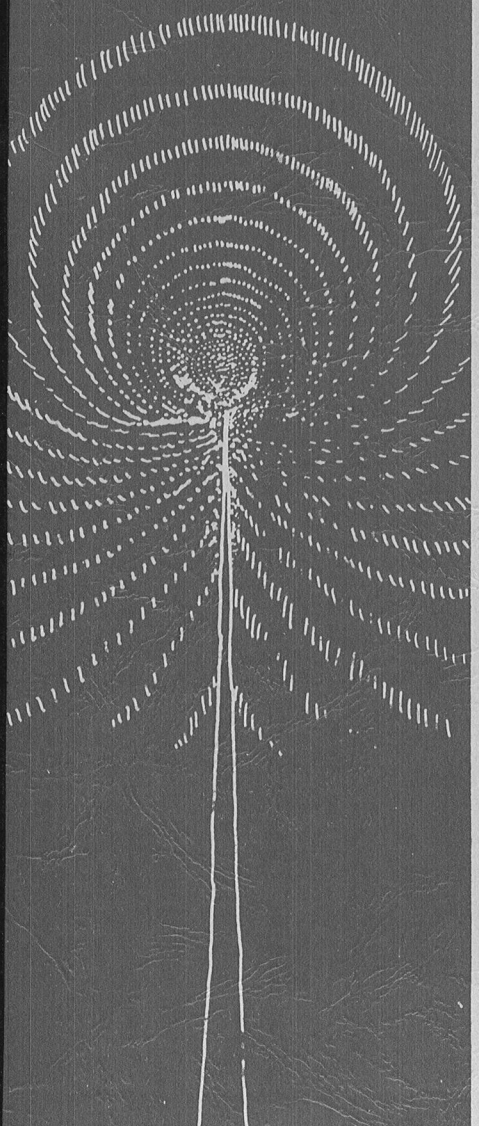

LEHIGH UNIVERSITY

OFF-AXIS IMPACT OF UNIDIRECTIONAL COMPOSITES WITH CRACKS: DYNAMIC STRESS INTENSIFICATION

BY

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MATERIALS AND STRUCTURES DIVISION
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16. Abstract

The dynamic response of unidirectional composites under off-axis (angle loading) impact is analyzed by assuming that the composite contains an initial flaw in the matrix material. The analytical method utilizes Fourier transform for the space variable and Laplace transform for the time variable. The off-axis impact is separated into two parts; one being symmetric and the other skew-symmetric with reference to the crack plane. Transient boundary conditions of normal and shear tractions are applied to a crack embedded in the matrix of the unidirectional composite. The two boundard conditions are solved independently and the results superimposed. Mathematically, these conditions reduce the problem to a system of dual integral equations which are solved in the Laplace transform plane for the transform of the dynamic stress intensity factor. The time inversion is carried out numerically for various combinations of the material properties of the composite and the results are displayed graphically.

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FOREWORD

This research report is concerned with the dynamic response of unidirectional composites under off-axis impact and represents a portion of the work performed for the NASA-Lewis Research Center in Cleveland, Ohio for the period February 13, 1978 through February 12, 1979 under Grant NSG 3179 with the Institute of Fracture and Solid Mechanics at Lehigh University. The Principal Investigator of the project is Professor George C. Sih and the Associate Investigator is Dr. E. P. Chen who has since left Lehigh University and joined the Sandia Laboratory in New Mexico. The authors are grateful to the NASA Project Manager, Dr. Christos C. Chamis who has carefully reviewed this report and provided a number of concrete suggestions.

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a	- half of the crack length
$A(s,p), B(s,p)$	- unknowns in dual integral equations
$A^{(i)}, B^{(i)}, C^{(i)}$	- coefficients for transfer of solution, function of (s,p)
Br	- Bromwich contour in the complex p-plane
c_{1j}, c_{2j}	- dilatational and shear wave speeds for medium j
$f^*(p)$	- Laplace transform of f(t)
$f^c(s)$	- cosine transform of f(x)
$f^s(s)$	- sine transform of f(x)
$(f)_j$	- indicates that f is evaluated in medium j
$F_I(s,p), F_{II}(s,p)$	- kernels in dual integral equations
h	- half of the thickness of the layer
$H(t)$	- Heaviside unit step function
$J_o(x)$	- Bessel function of order 0
$k_1(t), k_2(t)$	- dynamic stress intensity factors
$K_I(\xi,n,p)$ $K_{II}(\xi,n,p)$	- kernels in Fredholm integral equations
$P_n(x)$	- Legendre polynomial
r_1, θ_1	- crack tip polar coordinates
t	- time
u_x, u_y	- displacement components
x,y,z	- rectangular coordinates - crack lies in the xz-plane
$\alpha^{(i)}$	- functions of (p,s) through γ_{ij}
$\beta^{(i)}, \Delta_o$	- functions of (p,s) through $\alpha^{(i)}$
γ_{ij}	- exponents for transform of solution, functions of (p,s)
δ	- step size for numerical inversion of Laplace transforms
Δ_I, Δ_{II}	- functions of (p,s) through $\beta^{(i)}$

κ_i - parameters of dual integral equations

λ_1, λ_2 - Lamé coefficient

μ_1, μ_2 - shear modulus

ν_1, ν_2 - Poisson's ratio

ρ_1, ρ_2 - mass density

σ_o - suddenly applied normal stress

$\sigma(t)$ - time-dependent remote applied stress

$\sigma_x, \sigma_y, \sigma_z, \tau_{xy}$ - stress components for plane strain

τ_o - suddenly applied shear stress

ϕ_j, ψ_j - scalar potentials for medium j

$\Phi_I^*(\xi,p), \Phi_{II}^*(\xi,p)$ - unknowns in Fredholm integral equations

∇^2 - Laplacian operator

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ABSTRACT

The dynamic response of unidirectional composites under off-axis (angle loading) impact is analyzed by assuming that the composite contains an initial flaw in the matrix material. Because of the complexities that arise from the interaction of waves scattered by the crack with those reflected by the interfaces within the composite, dynamic analyses of composites with cracks have been treated only for a few simple cases. One of the objectives of the present work is to develop an effective analytical method for determining dynamic stress solutions. This will not only lead to an in-depth understanding of the failure of composites due to impact but also provide reliable solutions that can guide the development of numerical methods.

The analysis method utilizes Fourier transform for the space variable and Laplace transform for the time variable. The time-dependent angle loading is

*This work was completed during Dr. Chen's tenure at Lehigh University.

separated into two parts: one being symmetric and the other skew-symmetric with reference to the crack plane. By means of superposition, the transient boundary conditions consist of applying normal and shear tractions to a crack embedded in the matrix of the unidirectional composite. Mathematically, these conditions reduce the problem to a system of dual integral equations which are solved in the Laplace transform plane for the transform of the dynamic stress intensity factor. The time inversion is carried out numerically for various combinations of the material properties of the composite and the results are displayed graphically.

INTRODUCTION

Past work on the development of high performance composite materials was mainly concerned with achieving high strength and modulus. This requirement alone, however, may result in a composite that is excessively brittle and lacks the ability to resist impact loading. The energy absorption or toughness of the composite is also an important property that must be accounted for in addition to strength and stiffness.

The concept of fracture toughness has mostly been applied to homogeneous isotropic materials [1] based on the linear fracture mechanics theories such as those advanced by Griffith, Irwin and others. These theories, developed for single-phase materials, have had limited success in characterizing the fracture behavior of composites which are inherently nonhomogeneous and anisotropic. This is mainly because the fracture modes in composites are multi-facet and can include interface failure, fiber breaking, matrix fracture, etc. The individual contribution of each of these failure modes is not clearly accounted for and/or not related to the critical failure load. As a result, large discrepancies

between the theory and experiment can result.

A study on the selection of appropriate mathematical models for different unidirectional composite systems was made [2] in the case of static loading. Many of the assumptions in [2] will also be used in the dynamic problem treated here. One of them is the existence of inherent flaws or cracks which are the sites of failure initiation.

Analytical investigation of the fracture of fibrous composite materials subjected to impact loading has been meager because the elastodynamic stress analysis involves numerous parameters and is enormously complex. This is necessitated by the complex nature of the dynamic load transfer characteristics in composites containing initial imperfections such as flaws or cracks. The stress wave solution is not only time-dependent but it interacts with the material properties of the constituents of the composite and the various geometric parameters. The influence of these parameters will be analyzed in this impact study with particular emphasis placed on determining the dynamic stress intensity factors k_1 and k_2 arising from normal and shear loading. Their combination (off-axis or angle loading) determines the response to loading of a more general nature and reflects the energy absorption property of the composite. Several examples of how k_1 and k_2 can be combined to predict crack behavior in dynamic stress fields are found in [3]. The question of whether there is the need of how to define a dynamic fracture toughness parameter differing from its corresponding static value has been the subject of many past and present debates within the fracture mechanics community. Thus far, no general agreement has been achieved.

This report is concerned with dynamic fracture analysis and, particularly, with the development of an analytical method for obtaining effective dynamic stress solutions to unidirectional composites with cracks embedded in the matrix. Other possible failure modes will be dealt with in future reports. Effective stress solutions for k_1 and k_2 are essential as they are the prerequisites for formulating failure criteria and guiding the development of numerical procedures.

ANGLE CRACK UNDER IMPACT

Figure 1(a) considers a crack in a layer of matrix material of thickness 2h. The composite is reinforced by unidirectional fibers that are aligned parallel with one another and make an angle with the time-dependent applied stress $\sigma(t)$. Without serious loss in generality, the composite is assumed to be modeled by a layer of cracked material with elastic properties μ_1, ν_1 and ρ_1 sandwiched in between two dissimilar media with properties μ_2, ν_2 and ρ_2, Figure 1(b). The number of layers surrounding the cracked layer is reasonably large so that the average shear modulus μ_2, Poisson's ratio ν_2 and mass density ρ_2 can be used.

The basic two-dimensional elastodynamic equations in the theory of elasticity can be expressed in terms of two scalar potentials $\phi_j(x,y,t)$ and $\psi_j(x,y,t)$ where $i,j = 1,2$ with 1 and 2 referring to the cracked layer and the surrounding material, respectively. In terms of the Lamé coefficients λ_j and μ_j, the dynamic stress components are

$$(\sigma_x)_j = \lambda_j \nabla^2 \phi_j + 2\mu_j \left(\frac{\partial^2 \phi_j}{\partial x^2} + \frac{\partial^2 \psi_j}{\partial x \partial y}\right)$$

$$(\sigma_y)_j = \lambda_j \nabla^2 \phi_j + 2\mu_j \left(\frac{\partial^2 \phi_j}{\partial y^2} - \frac{\partial^2 \psi_j}{\partial x \partial y}\right) \tag{1}$$

$$(\sigma_z)_j = \frac{\lambda_j}{2} \left(\frac{\lambda_j + 2\mu_j}{\lambda_j + \mu_j}\right) \nabla^2 \phi_j$$

$$(\tau_{xy})_j = \mu_j \left(2 \frac{\partial^2 \phi_j}{\partial x \partial y} + \frac{\partial^2 \psi_j}{\partial y^2} - \frac{\partial^2 \psi_j}{\partial x^2}\right)$$

where $\nabla^2 = \partial^2/\partial x^2 + \partial^2/\partial y^2$ and the thickness shear stresses are assumed to vanish. The corresponding in-plane displacements are given by

$$(u_x)_j = \frac{\partial \phi_j}{\partial x} + \frac{\partial \psi_j}{\partial y}$$

$$(u_y)_j = \frac{\partial \phi_j}{\partial y} - \frac{\partial \psi_j}{\partial x} \tag{2}$$

Under plane strain, the material elements are constrained in the z-direction. The governing differential equations can then be obtained from the equations of motion:

$$\nabla^2 \phi_j = \frac{1}{c_{1j}^2} \frac{\partial^2 \phi_j}{\partial t^2}$$

$$\nabla^2 \psi_j = \frac{1}{c_{2j}^2} \frac{\partial^2 \psi_j}{\partial t^2} \tag{3}$$

in which c_{1j} and c_{2j} are the dilatational and shear wave velocities defined as

$$c_{1j} = \left(\frac{\lambda_j+2\mu_j}{\rho_j}\right)^{1/2}, \quad c_{2j} = \left(\frac{\mu_j}{\rho_j}\right)^{1/2} \tag{4}$$

The problem involves the determination of the potentials $\phi_j(x,y,t)$ and $\psi_j(x,y,t)$ in equations (3) from the transient boundary conditions of the crack problem.

The analysis may be simplified considerably if the problem is separated into two parts. The first concerns with normal stresses applied to the crack such that symmetry prevails about the x-axis in Figure 1(b) while the second deals with shear surface tractions so that the problem is skew-symmetric with reference to the x-axis. Both of these problems will be presented separately.

NORMAL IMPACT

Let the composite body be initially at rest such that the stresses are zero everywhere. Suddenly, at t=0, a normal stress of magnitude $-\sigma_0$ is applied to the top and bottom crack surfaces in Figure 1(b) and kept on the crack of length 2a thereafter. Referring to the set of axes x and y that are placed parallel and normal to the line crack, the following conditions are prescribed:

$$(\sigma_y)_1(x,o,t) = -\sigma_o H(t); \quad (\tau_{xy})_1(x,o,t) = 0, \quad 0\leq x<a; \quad t>0 \tag{5}$$

where H(t) is the Heaviside unit step function. The symmetry conditions about the axis y=0 are enforced by noting

$$(u_y)_1(x,o,t) = 0; \quad (\tau_{xy})_1(x,o,t) = 0, \quad x\geq a; \quad t>0 \tag{6}$$

Perfect bonding will be assumed along the interfaces between material 1 and material 2. This requires the stresses and displacements to be continuous across $y = \pm h$. On account of symmetry, only the upper half plane $y \geq 0$ need to be considered, i.e.,

$$(\sigma_y)_1(x,h,t) = (\sigma_y)_2(x,h,t)$$
$$(\tau_{xy})_1(x,h,t) = (\tau_{xy})_2(x,h,t) \tag{7}$$

and for the stresses and

$$(u_x)_1(x,h,t) = (u_x)_2(x,h,t)$$
$$(u_y)_1(x,h,t) = (u_y)_2(x,h,t) \tag{8}$$

for the displacements.

Dual integral equations. It is convenient at this point to apply the Laplace transform to the time variable t which corresponds to p in the transformed plane. Consider the standard Laplace transform on f(t):

$$f^*(p) = \int_0^\infty f(t)\, e^{-pt} dt \tag{9}$$

whose inversion is

$$f(t) = \frac{1}{2\pi i} \int_{Br} f^*(p)\, e^{pt} dp \tag{10}$$

in which Br stands for the Bromwich path of integration. The application of equation (9) to equations (3) yields

$$\nabla^2\phi_j^* = \frac{p^2}{c_{1j}^2}\phi_j^*$$

$$\nabla^2\psi_j^* = \frac{p^2}{c_{2j}^2}\psi_j^* \tag{11}$$

Again, the condition of symmetry requires only the consideration of x and y in the first quadrant. The Fourier cosine and sine transforms defined by

$$f^C(s) = \int_0^\infty f(x)\cos(sx)dx$$

$$f(x) = \frac{2}{\pi}\int_0^\infty f^C(s)\cos(sx)ds \tag{12}$$

and

$$f^S(s) = \int_0^\infty f(x)\sin(sx)dx$$

$$f(x) = \frac{2}{\pi}\int_0^\infty f^S(s)\sin(sx)ds \tag{13}$$

are now applied to the space variable x. This simplifies equations (3) to a set of ordinary differential equations which can be solved giving

$$\phi_1^*(x,y,p) = \frac{2}{\pi}\int_0^\infty [A^{(1)}(s,p)e^{-\gamma_{11}y} + A^{(2)}(s,p)e^{\gamma_{11}y}]\cos(sx)ds$$

$$\psi_1^*(x,y,p) = \frac{2}{\pi}\int_0^\infty [B^{(1)}(s,p)e^{-\gamma_{21}y} + B^{(2)}(s,p)e^{\gamma_{21}y}] \sin(sx)ds \tag{14}$$

for the cracked matrix and

$$\phi_2^*(x,y,p) = \frac{2}{\pi}\int_0^\infty C^{(1)}(s,p)e^{-\gamma_{12}y} \cos(sx)ds$$

$$\psi_2^*(s,y,p) = \frac{2}{\pi}\int_0^\infty C^{(2)}(s,p)e^{-\gamma_{22}y} \sin(sx)ds \tag{15}$$

for the averaged fiber-matrix material. In equations (14) and (15), the quantities γ_{1j} and γ_{2j} are given by

$$\gamma_{1j} = (s^2 + \frac{p^2}{c_{1j}^2})^{1/2}, \quad \gamma_{2j} = (s^2 + \frac{p^2}{c_{2j}^2})^{1/2} \tag{16}$$

The functions $A^{(1)}$, $A^{(2)}$, $B^{(1)}$,---, $C^{(2)}$ in equations (14) and (15) are determined from the transient boundary conditions. To this end, equations (5) and (6) will be written in the Laplace transform plane:

$$(\sigma_y^*)_1(x,o,p) = -\frac{\sigma_o}{p}; \quad (\tau_{xy}^*)_1(x,o,p) = 0, \quad 0 \leq x < a \tag{17}$$

and

$$(u_y^*)_1(x,o,p) = 0; \quad (\tau_{xy}^*)_1(x,o,p) = 0, \quad x \geq a \tag{18}$$

In the same way, equations (7) become

$$(\sigma_y^*)_1(x,h,p) = (\sigma_y^*)_2(x,h,p)$$

$$(\tau_{xy}^*)_1(x,h,p) = (\tau_{xy}^*)_2(x,h,p) \tag{19}$$

and equations (8) take the forms

$$(u_x^*)_1(x,h,p) = (u_x^*)_2(x,h,p)$$

$$(u_y^*)_1(x,h,p) = (u_y^*)_2(x,h,p) \tag{20}$$

The stresses and displacements in equations (1) and (2) may also be transformed into the Laplace transform plane. Without going into details, the appropriate Laplace transform of the stress and displacement expressions in equations (17) to (20) may be used to satisfy all of the necessary boundary, symmetry and continuity conditions. This leads to the following set of dual integral equations:

$$\int_0^\infty A(s,p)\cos(sx)ds = 0, \quad x \geq a$$

$$\int_0^\infty sF_I(s,p)\,A(s,p)\cos(sx)ds = -\frac{\pi\sigma_o}{4\mu_1 p(1-\kappa_1^2)}, \quad x<a \tag{21}$$

in which $F_I(s,p)$ stands for the known function

$$F_I(s,p) = \frac{1}{s(1-\kappa_1^2)\Delta_I}\{[\tfrac{1}{4}(s^2+\gamma_{21}^2)^2 - s^2\gamma_{11}\gamma_{21}][\beta^{(2)} - \beta^{(3)}e^{-2(\gamma_{11}+\gamma_{21})h}]$$

$$+ s(s^2+\gamma_{21}^2)[\gamma_{21}(\beta^{(1)}\beta^{(4)} - \beta^{(2)}\beta^{(3)}) - \gamma_{11}]e^{-(\gamma_{11}+\gamma_{21})h}$$

$$+ [\tfrac{1}{4}(s^2+\gamma_{21}^2)^2 + s^2\gamma_{11}\gamma_{21}][\beta^{(4)}e^{-2\gamma_{21}h} - \beta^{(1)}e^{-2\gamma_{11}h}]\} \qquad (22)$$

while the quantities κ_1 and Δ_I are defined as

$$\kappa_1 = (c_{21}/c_{11})^{1/2}$$

$$\Delta_I = \frac{p^2}{2c_{21}^2}\gamma_{11}[\beta^{(2)} + \beta^{(3)}\, e^{-2(\gamma_{11}+\gamma_{21})h} + \beta^{(4)}\, e^{-2\gamma_{21}h} \qquad (23)$$

$$+ \beta^{(1)}\, e^{-2\gamma_{11}h}]$$

such that $\beta^{(1)}$, $\beta^{(2)}$,---, $\beta^{(4)}$ are given by

$$\beta^{(1)} = (\alpha^{(3)}\alpha^{(6)} - \alpha^{(2)}\alpha^{(7)})/\Delta_o;\ \beta^{(2)} = (\alpha^{(4)}\alpha^{(6)} - \alpha^{(2)}\alpha^{(8)})/\Delta_o$$

$$\beta^{(3)} = (\alpha^{(1)}\alpha^{(7)} - \alpha^{(3)}\alpha^{(5)})/\Delta_o;\ \beta^{(4)} = (\alpha^{(1)}\alpha^{(8)} - \alpha^{(4)}\alpha^{(5)})/\Delta_o \qquad (24)$$

where Δ_o is

$$\Delta_o = \alpha^{(1)}\alpha^{(6)} - \alpha^{(2)}\alpha^{(5)} \qquad (25)$$

The quantities $\alpha^{(1)}$, $\alpha^{(2)}$,---, $\alpha^{(8)}$ in equations (25) are complicated functions of s, p and the material constants. They are given by equations (I.1) in Appendix I.

The problem is now reduced to finding the single unknown A(s,p) governed by equations (21). Once A(s,p) is known, the functions $A^{(1)}$, $A^{(2)}$,---, $C^{(2)}$ that are required in equations (14) and (15) for the Laplace transform of the potentials $\phi_j^*(x,y,p)$ and $\psi_j^*(x,y,p)$ can be obtained from equations (I.2) outlined in Appendix I. What remains is the determination of a solution for the dual integral equations (21). This will be accomplished with the help of a method by Copson [5] which has been used by Chen and Sih [6] for solving dynamic crack problems involving single-phase homogeneous materials. Following the details in [5,6], it can be shown that

$$A(s,p) = -\frac{\pi\sigma_0 a^2}{4\mu_1 p(1-\kappa_1^2)} \int_0^1 \sqrt{\xi}\, \Phi_I^*(\xi,p)\, J_0(sa\xi)d\xi \tag{26}$$

is a solution of equations (21) with J_0 being the zero order Bessel function of the first kind. The function $\Phi_I^*(\xi,p)$ is calculated numerically from a Fredholm integral equation of the second kind:

$$\Phi_I^*(\xi,p) + \int_0^1 \Phi_I^*(\eta,p)\, K_I(\xi,\eta,p)d\eta = \sqrt{\xi} \tag{27}$$

whose kernel

$$K_I(\xi,\eta,p) = \sqrt{\xi\eta} \int_0^\infty s[F_I(\tfrac{s}{a}, p) - 1]\, J_0(s\xi)\, J_0(s\eta)ds \tag{28}$$

is symmetric in ξ and η.

Mode I dynamic stress intensity factor. The transmission of the time-dependent load to the vicinity of the crack tip can be best described by the intensification of the local stresses. A quantity that has been used widely in

the static theory of fracture mechanics is the "stress intensity factor" which can be extracted from the asymptotic expansions of the stresses near the crack tip. Referring to Figure 2, let r_1 and θ_1 be a set of local polar coordinates measured from the right hand crack tip located at x=a and y=0 in the matrix material, the singular character of the dynamic stresses is described only by the space variables and hence can be more easily determined in the Laplace transform domain. This observation was first made by Sih, Ravera and Embley [7]. Following their procedure, the local stresses in terms of r_1 and θ_1 are found:

$$
\begin{aligned}
(\sigma_x^*)_1(r_1,\theta_1,p) &= \frac{k_1^*(p)}{\sqrt{2r_1}} \cos\frac{\theta_1}{2}\left(1 - \sin\frac{\theta_1}{2}\sin\frac{3\theta_1}{2}\right) + O(r_1^\circ) \\
(\sigma_y^*)_1(r_1,\theta_1,p) &= \frac{k_1^*(p)}{\sqrt{2r_1}} \cos\frac{\theta_1}{2}\left(1 + \sin\frac{\theta_1}{2}\sin\frac{3\theta_1}{2}\right) + O(r_1^\circ) \\
(\sigma_z^*)_1(r_1,\theta_1,p) &= \frac{k_1^*(p)}{\sqrt{2r_1}} 2\nu_1 \cos\frac{\theta_1}{2} + O(r_1^\circ) \\
(\tau_{xy}^*)_1(r_1,\theta_1,p) &= \frac{k_1^*(p)}{\sqrt{2r_1}} \sin\frac{\theta_1}{2}\cos\frac{\theta_1}{2}\cos\frac{3\theta_1}{2} + O(r_1^\circ)
\end{aligned}
\tag{29}
$$

Only the dynamic stress intensity factor, $k_1^*(p)$, in equations (29) need to be inverted to real time t:

$$
k_1^*(p) = \frac{\Phi_I^*(1,p)}{p}\sigma_o\sqrt{a} \tag{30}
$$

where the function $\Phi_I^*(1,p)$ is found from $\Phi_I^*(\xi,p)$ by letting the nondimensional parameter ξ=1 representing the crack tip location. The functional dependence

of the stresses in r_1 and θ_1 as shown by equations (29) reveals that the dynamic stresses also possess the inverse square root singularity in terms of r_1 and that the angular distribution in θ, is the same as the case for static loading.

Applying the Laplace inversion formula in equation (10) to (30) renders the factor $k_1(t)$ as a function of time, i.e.,

$$k_1(t) = \frac{\sigma_o\sqrt{a}}{2\pi i} \int_{Br} \frac{\Phi_I^*(1,p)}{p} e^{pt} dp \tag{31}$$

It is apparent that $\Phi_I^*(1,p)$ must be first known before the integration of equation (31) can be performed. Refer to Appendix II for a detailed account of the procedure used for evaluating equation (31). Three different sets of $\Phi_I^*(1,p)$ values are plotted against the dimensionless Laplace transform wave number c_{21}/pa. They are given in Figures 3 to 5 for $\rho_1 = \rho_2$ and $\nu_1 = \nu_2 = 0.29$ while the ratios a/h and μ_2/μ_1 are varied. In general, all the curves tend to rise quickly and then flatten out. It would be more meaningful to discuss the influence of a/h and μ_2/μ_1 on the stress intensity factor $k_1(t)$.

Figures 6 to 8 display the normalized stress intensity factor $k_1(t)/\sigma_o\sqrt{a}$ as a function of $c_{21}t/a$. In Figure 6, the crack length to layer thickness ratio, a/h, is fixed at unity while the shear moduli ratio, μ_2/μ_1 is increased from 0.1 to 10.0 as indicated. The $k_1(t)$ factor is oscillatory in nature reaching a peak and then decreases in magnitude as time increases. The oscillation is more pronounced when the shear modulus of the cracked material is greater than that of the surrounding material, i.e., $\mu_1 > \mu_2$. The values of $k_1(t)$ decrease below those of the corresponding homogeneous case, $\mu_1 = \mu_2$, solved previously by Chen and Sih [6] when $\mu_1 < \mu_2$. The influence of a/h on $k_1(t)$ is exhibited in Figures 7 and 8 for the two cases of $\mu_2/\mu_1 = 0.1$ and 10.0, respectively. For $\mu_2/\mu_1 = 0.1$

in Figure 7, a decrease in a/h tends to lower the stress intensity factor. Observed also is a small step in the curve for a/h = 2.0 and small time t. This corresponds to the reflection of elastic waves from the material interface. The size and time scale are such that this effect showed up quantitatively in the graph while the same effect was not noticeable in the other curves. For the smaller ratios of a/h such as 0.5 and 1.0, the crack tips are further away from the interface and the influence of the reflected waves are not as pronounced. The opposite trend is observed in Figure 8 for $\mu_2/\mu_1 = 10.0$. When the outer material is more rigid than that of the center layer, $k_1(t)$ tends to increase in magnitude as a/h is decreased. Again, a distinct step in the curve for a/h = 2.0 is seen for small time t. As time increases, all of the results here reduce to the corresponding static solutions of Hilton and Sih [8].

General loading. If the normal stress applied to the crack surface is not constant in magnitude but may vary as a function of x, then the dynamic stress intensity factor can be obtained by adding a sequence of solutions corresponding to step loadings with different stress levels σ_0, σ_1, etc. In other words, the general loading $\sigma(t)$ may be considered as the sum:

$$\sigma(t) = \sigma_0 H(t_0) + \sigma_1 H(t_1) + \sigma_2 H(t_2) + \ldots \tag{32}$$

This is illustrated graphically in Figure 9. From equations (31) and (32), the factor $k_1(t)$ that corresponds to $\sigma(t)$ may be written down immediately as follows:

$$k_1(t) = \frac{1}{2\pi i} [\sigma_0 H(t_0) + \sigma_1 H(t_1) + \ldots] \int_{Br} \frac{\Phi_I^*(1,p)}{p} e^{pt} dp \tag{33}$$

Equation (33) may be used to derive $k_1(t)$ for any time-dependent normal surface tractions which in turn can also simulate any loadings that are applied at dis-

tances away from the crack by means of the principle of superposition.

SHEAR IMPACT

Suppose that the crack in Figure 1(b) is now sheared suddenly by a pair of shear stresses of magnitude $-\tau_o$ such that the upper and lower crack surfaces move in the opposite direction. This creates a deformation field that is skew-symmetric with respect to the y=0 plane. Following the footstep laid out in the previous example on normal impact, the Laplace transform of the transient boundary conditions on the x-axis inside the crack are

$$(\tau^*_{xy})_1(x,o,p) = -\frac{\tau_o}{p};\ (\sigma^*_y)_1(x,o,p) = 0,\ 0\leq x<a \tag{34}$$

and the skew-symmetric conditions outside the crack are given by

$$(u^*_x)_1(x,o,p) = 0;\ (\sigma^*_y)_1(x,o,p) = 0,\ x\geq a \tag{35}$$

Continuity of the stresses across y=h is expressed by

$$\begin{aligned}(\sigma^*_y)_1(x,h,p) &= (\sigma^*_y)_2(x,h,p)\\ (\tau^*_{xy})_1(x,h,p) &= (\tau^*_{xy})_2(x,h,p)\end{aligned} \tag{36}$$

while the displacements are also required to be continuous, i.e.,

$$\begin{aligned}(u^*_x)_1(x,h,p) &= (u^*_x)_2(x,h,p)\\ (u^*_y)_1(x,h,p) &= (u^*_y)_2(x,h,p)\end{aligned} \tag{37}$$

Integral representations. Under the above considerations, the following wave potentials $\phi_j^*(x,y,p)$ and $\psi_j^*(x,y,p)$ are selected:

$$\phi_1^*(x,y,p) = \frac{2}{\pi}\int_0^\infty [A^{(1)}(s,p)e^{-\gamma_{11}y} + A^{(2)}(s,p)e^{\gamma_{11}y}] \sin(sx)ds$$

$$\psi_1^*(x,y,p) = \frac{2}{\pi}\int_0^\infty [B^{(1)}(s,p)e^{-\gamma_{21}y} + B^{(2)}(s,p)e^{\gamma_{21}y}] \cos(sx)ds \tag{38}$$

for the cracked layer and

$$\phi_2^*(x,y,p) = \frac{2}{\pi}\int_0^\infty C^{(1)}(s,p)e^{-\gamma_{12}y} \sin(sx)ds$$

$$\psi_2^*(x,y,p) = \frac{2}{\pi}\int_0^\infty C^{(2)}(s,p)e^{-\gamma_{22}y} \cos(sx)ds \tag{39}$$

for the outside material.

Equations (38) and (39) may now be substituted into the Laplace transform of the stresses and displacements in equations (1) and (2). Making use of the conditions in equations (34) to (37), the solution can be expressed in terms of the functions $A^{(1)}$, $A^{(2)}$,---, $C^{(2)}$ which are related to a single unknown B(s,p) as shown by equations (III.1) in Appendix III. The function B(s,p) is governed by the system of dual integral equations

$$\int_0^\infty B(s,p) \cos(sx)ds = 0, \quad x \geq a$$

$$\int_0^\infty sF_{II}(s,p)\, B(s,p) \cos(sx)ds = \frac{\pi\tau_0}{4\mu_1 p(1-\kappa_1^2)}, \quad x<a \tag{40}$$

The function $F_{II}(s,p)$ is related to $F_I(s,p)$ in equation (22) as

$$F_{II}(s,p) = \frac{\Delta_I}{\Delta_{II}} F_I(s,p) \tag{41}$$

where

$$\Delta_{II} = \frac{p^2}{2c_{21}^2} \gamma_{21}[\beta^{(2)} + \beta^{(3)} e^{-2(\gamma_{11}+\gamma_{21})h} - \beta^{(4)} e^{-2\gamma_{21}h} - \beta^{(1)} e^{-2\gamma_{11}h}] \tag{42}$$

The other parameters such as κ_1, Δ_I, $\beta^{(1)}$, $\beta^{(2)}$, etc., are the same as those defined earlier for the case of normal impact.

A solution to equations (40) is again found by application of the Copson's method [5]:

$$B(s,p) = \frac{\pi\tau_o a^2}{4\mu_1 p(1-\kappa_1^2)} \int_0^1 \sqrt{\xi}\, \Phi_{II}^*(\xi,p)\, J_o(sa\xi)d\xi \tag{43}$$

provided that $\Phi_{II}^*(\xi,p)$ satisfies a Fredholm integral equation of the second kind:

$$\Phi_{II}^*(\xi,p) + \int_0^1 \Phi_{II}^*(\eta,p)\, K_{II}(\xi,\eta,p)d\eta = \sqrt{\xi} \tag{44}$$

whose kernel $K_{II}(\xi,\eta,p)$ takes the form

$$K_{II}(\xi,\eta,p) = \sqrt{\xi\eta} \int_o^\infty s[F_{II}(\tfrac{s}{a}, p) - 1]\, J_o(s\xi)\, J_o(s\eta)ds \tag{45}$$

Mode II dynamic stress intensity factor. As in the case of Mode I, the asymptotic expressions of the dynamic stresses in the Laplace transform plane are first obtained in terms of r_1 and θ_1 defined in Figure 2. The results are

$$(\sigma_x^*)_1(r_1,\theta_1,p) = -\frac{k_2^*(p)}{\sqrt{2r_1}}\sin\frac{\theta_1}{2}\left(2+\cos\frac{\theta_1}{2}\cos\frac{3\theta_1}{2}\right)+O(r_1^\circ)$$

$$(\sigma_y^*)_1(r_1,\theta_1,p) = \frac{k_2^*(p)}{\sqrt{2r_1}}\sin\frac{\theta_1}{2}\cos\frac{\theta_1}{2}\cos\frac{3\theta_1}{2}+O(r_1^\circ) \qquad (46)$$

$$(\sigma_z^*)_1(r_1,\theta_1,p) = -\frac{k_2^*(p)}{\sqrt{2r_1}}2\nu_1\sin\frac{\theta_1}{2}+O(r_1^\circ)$$

$$(\tau_{xy}^*)_1(r_1,\theta_1,p) = \frac{k_2^*(p)}{\sqrt{2r_1}}\sin\frac{\theta_1}{2}\cos\frac{\theta_1}{2}\cos\frac{3\theta_1}{2}+O(r_1^\circ)$$

with $k_2^*(p)$ being the only quantity that depends on time through the parameter p:

$$k_2^*(p) = \frac{\Phi_{II}^*(1,p)}{p}\tau_o\sqrt{a} \qquad (47)$$

Equation (10) is then applied to invert the Laplace transform of the stress intensity factor in equation (47). This gives

$$k_2(t) = \frac{\tau_o\sqrt{a}}{2\pi i}\int_{Br}\frac{\Phi_{II}^*(1,p)}{p}e^{pt}dp \qquad (48)$$

in which $\Phi_{II}^*(1,p)$ is computed numerically from equation (44).

Figures 10 to 12 display the values of $\Phi_{II}^*(1,p)$ as a function of the normalized quantity c_{21}/pa for various values of a/h and μ_2/μ_1 while $\nu_1 = \nu_2 = 0.29$ and $\rho_1 = \rho_2$ are used for all cases. With a knowledge of $\Phi_{II}^*(1,p)$, the integral in equation (48) may be evaluated by a procedure outlined in Appendix II. In general, $k_2(t)$ increases with time reaching a maximum and then decreases to the static value for sufficiently large time. The trend is very similar to $k_1(t)$

for the case of normal impact in that a higher value of $k_2(t)$ is obtained when the modulus of the surrounding material is less than that of the cracked layer, i.e., $\mu_2/\mu_1 < 1$. Comparing the results in Figures 6 and 12, it is seen that for $\mu_2/\mu_1 < 1$, normal impact yields a higher crack tip stress intensity factor than shear impact, i.e., $k_1(t) > k_2(t)$. The opposite is observed when $\mu_2/\mu_1 > 1$, i.e., $k_2(t) > k_1(t)$. The curves in Figures 14 and 15 for $k_2(t)$ also show the absence of a small fluctuation for small time which was present in Figures 7 and 8 for $k_1(t)$. This is because the influence of the reflected incident shear wave from the interface is considerably weaker even for the ratio of a/h = 2.0.

CONCLUSION

As composite materials are currently being applied to major primary structure designs, it is necessary to have an in-depth understanding of the mechanical behavior of these materials, particularly with reference to the allowable applied load both statically and dynamically. This investigation is concerned with the dynamic stress distribution around a crack embedded in the matrix of a unidirectional composite. The time-dependent loading can be of a general nature applied in an arbitrarily direction with reference to the crack plane. For those composites which fail predominantly by matrix cracking under impact, the present results can be used effectively for determining the ability of the composite to absorb energy and to withstand load prior to total destruction.

The other modes of failure such as fiber breaking and/or debonding of fibers from matrix are not treated but may be significant in other composite systems. The redistribution of dynamic stresses in these cases may also be analyzed such that their individual contribution can be assessed quantitatively. These cases will be left for future investigations.

APPENDIX I: EXPRESSIONS FOR $\alpha^{(i)}$ AND $A^{(i)}(s,p),---, C^{(i)}(s,p)$ IN NORMAL LOADING

This section gives the expressions for $\alpha^{(1)}$, $\alpha^{(2)},---, \alpha^{(8)}$ in equations (25) in terms of the variables s, p and the material constants

$$\alpha^{(1)} = -s[(1 - \frac{\mu_2}{\mu_1})\gamma_{21} - \frac{\mu_2}{\mu_1}(\frac{p^2}{2c_{22}^2})(\frac{\gamma_{21}-\gamma_{22}}{s^2-\gamma_{12}\gamma_{22}})]$$

$$\alpha^{(2)} = s[(1 - \frac{\mu_2}{\mu_1})\gamma_{21} - \frac{\mu_2}{\mu_1}(\frac{p^2}{2c_{22}^2})(\frac{\gamma_{21}+\gamma_{22}}{s^2-\gamma_{12}\gamma_{22}})]$$

$$\alpha^{(3)} = \frac{1}{2}(s^2+\gamma_{21}^2) - \frac{\mu_2}{\mu_1}[s^2 + \frac{p^2}{2c_{22}^2}(\frac{s^2-\gamma_{11}\gamma_{22}}{s^2-\gamma_{12}\gamma_{22}})]$$

$$\alpha^{(4)} = \frac{1}{2}(s^2+\gamma_{21}^2) - \frac{\mu_2}{\mu_1}[s^2 + \frac{p^2}{2c_{22}^2}(\frac{s^2+\gamma_{11}\gamma_{22}}{s^2-\gamma_{12}\gamma_{22}})] \qquad (I.1)$$

$$\alpha^{(5)} = -\frac{1}{2}(s^2+\gamma_{21}^2) + \frac{\mu_2}{\mu_1}[s^2 + \frac{p^2}{2c_{22}^2}(\frac{s^2-\gamma_{12}\gamma_{21}}{s^2-\gamma_{12}\gamma_{22}})]$$

$$\alpha^{(6)} = -\frac{1}{2}(s^2+\gamma_{21}^2) + \frac{\mu_2}{\mu_1}[s^2 + \frac{p^2}{2c_{22}^2}(\frac{s^2+\gamma_{12}\gamma_{21}}{s^2-\gamma_{12}\gamma_{22}})]$$

$$\alpha^{(7)} = s[(1 - \frac{\mu_2}{\mu_1})\gamma_{11} - \frac{\mu_2}{\mu_1}(\frac{p^2}{2c_{22}^2})(\frac{\gamma_{11}-\gamma_{12}}{s^2-\gamma_{12}\gamma_{22}})]$$

$$\alpha^{(8)} = -s[(1 - \frac{\mu_2}{\mu_1})\gamma_{11} - \frac{\mu_2}{\mu_1}(\frac{p^2}{2c_{22}^2})(\frac{\gamma_{11}+\gamma_{12}}{s^2-\gamma_{12}\gamma_{22}})]$$

in which γ_{ij} is given by equations (16).

The functions $A^{(1)}$, $A^{(2)}$,---, $C^{(2)}$ are related to the single function $A(s,p)$ as follows:

$$A^{(1)}(s,p) = \frac{A(s,p)}{\Delta_I}\left[\frac{1}{2}(s^2+\gamma_{21}^2)(\beta^{(2)} + \beta^{(4)}\, e^{-2\gamma_{21}h}) - s\gamma_{11}\, e^{-(\gamma_{11}+\gamma_{21})h}\right]$$

$$A^{(2)}(s,p) = -\frac{A(s,p)}{\Delta_I}\left[s\gamma_{11}\, e^{-(\gamma_{11}+\gamma_{21})h} + \frac{1}{2}(s^2+\gamma_{21}^2)\, e^{-2\gamma_{11}h}\,(\beta^{(1)} + \beta^{(3)}\, e^{-2\gamma_{21}h})\right]$$

$$B^{(1)}(s,p) = \beta^{(1)}\, e^{-(\gamma_{11}-\gamma_{21})h}\, A^{(1)}(s,p) + \beta^{(2)}\, e^{(\gamma_{11}+\gamma_{21})h}\, A^{(2)}(s,p)$$

$$B^{(2)}(s,p) = \beta^{(3)}\, e^{-(\gamma_{11}+\gamma_{21})h}\, A^{(1)}(s,p) + \beta^{(4)}\, e^{(\gamma_{11}-\gamma_{21})h}\, A^{(2)}(s,p)$$

$$C^{(1)}(s,p) = \frac{e^{\gamma_{12}h}}{s^2-\gamma_{12}\gamma_{22}}\left[(s^2-\gamma_{11}\gamma_{22})\, e^{-\gamma_{11}h}\, A^{(1)}(s,p) + (s^2+\gamma_{11}\gamma_{22})\, e^{\gamma_{11}h}\, A^{(2)}(s,p) + s(\gamma_{21}-\gamma_{22})\, e^{-\gamma_{21}h}\, B^{(1)}(s,p) - s(\gamma_{21}+\gamma_{22})\, e^{\gamma_{21}h}\, B^{(2)}(s,p)\right] \tag{I.2}$$

$$C^{(2)}(s,p) = \frac{e^{\gamma_{22}h}}{s^2-\gamma_{12}\gamma_{22}}\left[s(\gamma_{11}-\gamma_{12})\, e^{-\gamma_{11}h}\, A^{(1)}(s,p) - s(\gamma_{11}+\gamma_{22})\, e^{\gamma_{11}h}\, A^{(2)}(s,p) + (s^2-\gamma_{12}\gamma_{21})\, e^{-\gamma_{21}h}\, B^{(1)}(s,p) + (s^2+\gamma_{12}\gamma_{21})\, e^{\gamma_{21}h}\, B^{(2)}(s,p)\right]$$

APPENDIX II: METHOD FOR EVALUATING THE DYNAMIC STRESS INTENSITY FACTOR EQUATION (31)

The integral in equation (31) is basically of the form

$$g(t) = \frac{1}{2\pi i} \int_{Br} \frac{f^*(1,p)}{p} e^{pt} dp \qquad (II.1)$$

The Bromwich path, Br, consists of an infinite line parallel to and to the right of the imaginary axis in the complex p-plane. The function* $f^*(1,p)$ is considered to be known for discrete values of p. There are a number of available methods for finding g(t) as a process in the Laplace inverse transform. The method adopted here can be found in [9,10].

The integral $f^*(1,p)/p$ in equation (II.1) is first evaluated at the points

$$p = (1+n)\delta, \quad n = 0,1,2,--- \qquad (II.2)$$

in which δ is a real and positive number. According to equations (9) and (10), $f^*(1,p)/p$ may be written as

$$\frac{f^*(1,p)}{p} = \int_0^\infty g(t)e^{-pt} dt \qquad (II.3)$$

The above infinite integral is now transformed to a finite integral on the interval [-1,1] by making the substitutions

---

*$f^*(1,p)$ stands for $\Phi_I^*(1,p)$ in normal impact and $\Phi_{II}^*(1,p)$ in shear impact and they are calculated from the Fredholm integral equations of the second kind, namely equations (27) and (44).

$$x = 2e^{-\delta t} - 1 \tag{II.4}$$

and

$$G(x) = g[t(x)] = g[-\frac{1}{\delta} \log(\frac{x+1}{2})] \tag{II.5}$$

Therefore, equation (II.3) becomes

$$\frac{f^*[1,(1+n)\delta]}{1+n} = \frac{1}{2^{n+1}} \int_{-1}^{1} (1+x)^n \, G(x)dx \tag{II.6}$$

in which G(x) can be expanded in series form consisting of Legendre polynomials $P_n(x)$ which are orthogonal on the interval [-1,1], i.e.,

$$G(x) = \sum_{i=0}^{\infty} C_i \, P_i(x) \tag{II.7}$$

Similarly, the function $(1+x)^n$ in equation (II.6) may also be expanded in the form

$$(1+x)^n = \sum_{i=0}^{n} D_i \, P_i(x) \tag{II.8}$$

such that

$$D_i = 2^n(2i+1) \, \frac{n(n-1)---[n-(i-1)]}{(n+1)(n+2)---(n+i+1)} \tag{II.9}$$

Putting equations (II.7) and (II.8) into (II.6) and applying the orthogonality conditions for the Legendre polynomials, the following sum is established:

$$\frac{f^*[1,(1+n)\delta]}{1+n} = \sum_{i=0}^{n} \frac{n(n-1)---[n-(i-1)]}{(n+1)(n+2)---(n+i+1)} C_i \qquad \text{(II.10)}$$

Thus the coefficients C_i may be found with C_0 given by

$$C_0 = f^*(1,\delta) \qquad \text{(II.11)}$$

For a finite number of N coefficients, a partial sum for G(x) in (II.7) is obtained and an approximate evaluation of g(t) can be made since from equation (II.5)

$$g(t) = \sum_{i=0}^{N-1} C_i P_i \, [2e^{-\delta t} - 1] \qquad \text{(II.12)}$$

The parameter δ is chosen such that g(t) is best described for the range of t considered.

APPENDIX III: EXPRESSIONS FOR $A^{(i)}(s,p)$,---, $C^{(i)}(s,p)$ IN SHEAR LOADING

In the skew-symmetric problem, the unknown functions in equations (38) and (39) can also be expressed in terms of a single unknown B(s,p) in accordance with the following relationships:

$$A^{(1)}(s,p) = -\frac{B(s,p)}{\Delta_{II}}\left[s\gamma_{21}(\beta^{(2)} - \beta^{(4)}e^{-2\gamma_{21}h}) + \frac{1}{2}(s^2+\gamma_{21}^2)e^{-(\gamma_{11}+\gamma_{21})h}\right]$$

$$A^{(2)}(s,p) = \frac{B(s,p)}{\Delta_{II}}\left[s\gamma_{21}e^{-2\gamma_{11}h}(\beta^{(1)} - \beta^{(3)}e^{-2\gamma_{21}h}) + \frac{1}{2}(s^2+\gamma_{21}^2)e^{-(\gamma_{11}+\gamma_{21})h}\right]$$

$$B^{(1)}(s,p) = -\beta^{(1)}e^{-(\gamma_{11}-\gamma_{21})h}A^{(1)}(s,p) - \beta^{(2)}e^{(\gamma_{11}+\gamma_{21})h}A^{(2)}(s,p)$$

$$B^{(2)}(s,p) = -\beta^{(3)}e^{-(\gamma_{11}+\gamma_{21})h}A^{(1)}(s,p) - \beta^{(4)}e^{(\gamma_{11}-\gamma_{21})h}A^{(2)}(s,p)$$

$$C^{(1)}(s,p) = \frac{e^{\gamma_{12}h}}{s^2-\gamma_{12}\gamma_{22}}\left[(s^2-\gamma_{11}\gamma_{22})e^{-\gamma_{11}h}A^{(1)}(s,p) + (s^2+\gamma_{11}\gamma_{22})e^{\gamma_{11}h}A^{(2)}(s,p) - s(\gamma_{21}-\gamma_{22})e^{-\gamma_{21}h}B^{(1)}(s,p) + s(\gamma_{21}+\gamma_{22})e^{\gamma_{21}h}B^{(2)}(s,p)\right]$$

$$C^{(2)}(s,p) = \frac{e^{\gamma_{22}h}}{s^2-\gamma_{21}\gamma_{22}} [s(\gamma_{12}-\gamma_{11})e^{-\gamma_{11}h} A^{(1)}(s,p) \tag{III.1}$$

$$+ s(\gamma_{12}+\gamma_{11})e^{\gamma_{11}h} A^{(2)}(s,p) + (s^2-\gamma_{21}\gamma_{12})e^{-\gamma_{21}h} B^{(1)}(s,p)$$

$$+ (s^2+\gamma_{21}\gamma_{12})e^{\gamma_{21}h} B^{(2)}(s,p)]$$

where Δ_{II} is given by equation (42).

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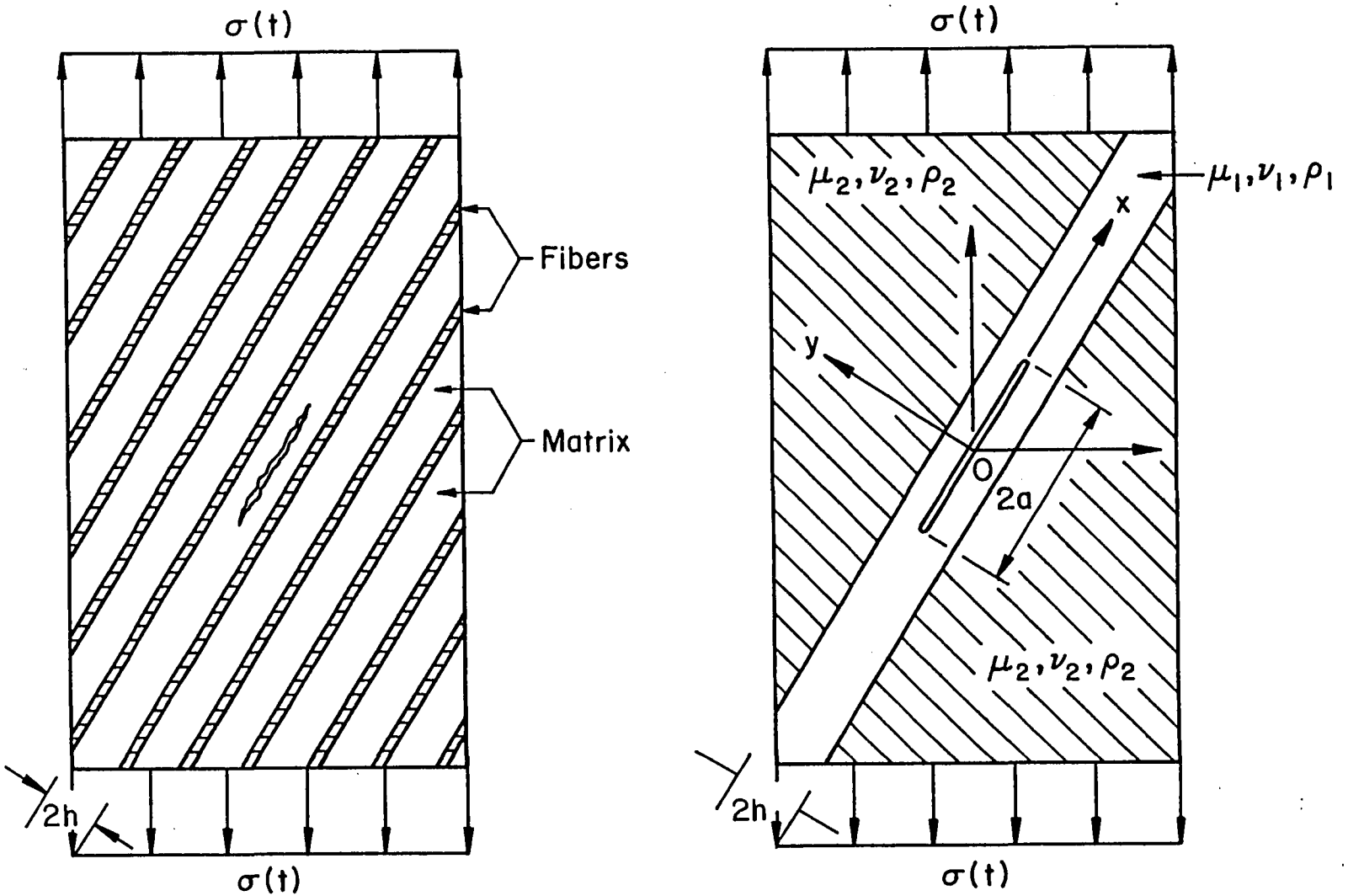

Figure 1. Fiber-reinforced unidirectional composite subjected to angle impact

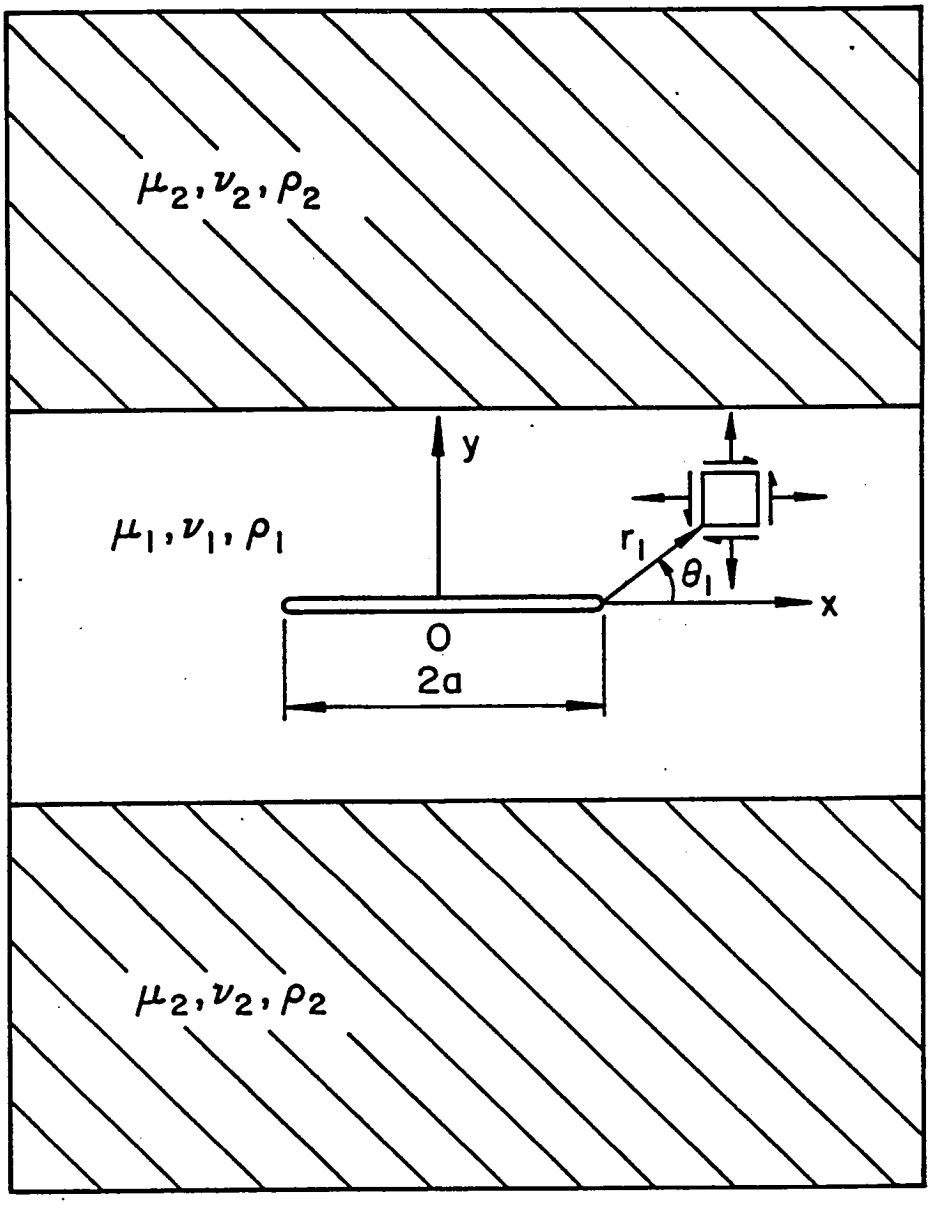

Figure 2. Stress element near crack in matrix of fiber-reinforced composite

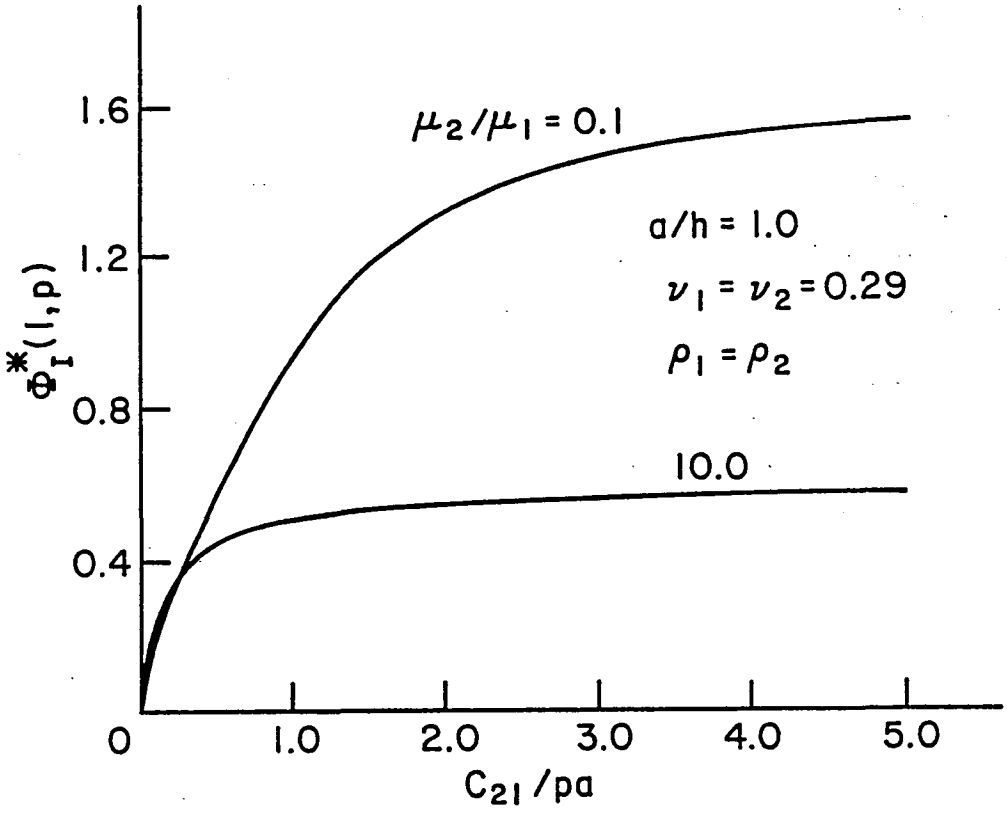

Figure 3. Variations of $\Phi_I^*(1,p)$ with c_{21}/pa for a/h = 1.0

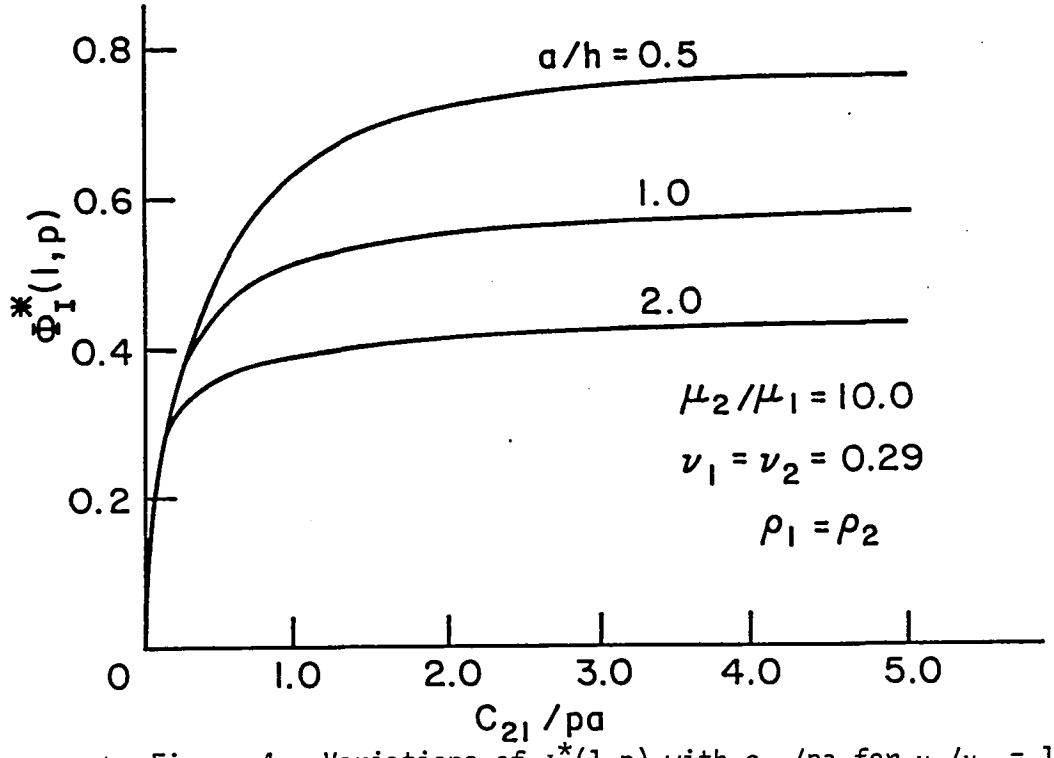

Figure 4. Variations of $\Phi_I^*(1,p)$ with c_{21}/pa for $\mu_2/\mu_1 = 10$

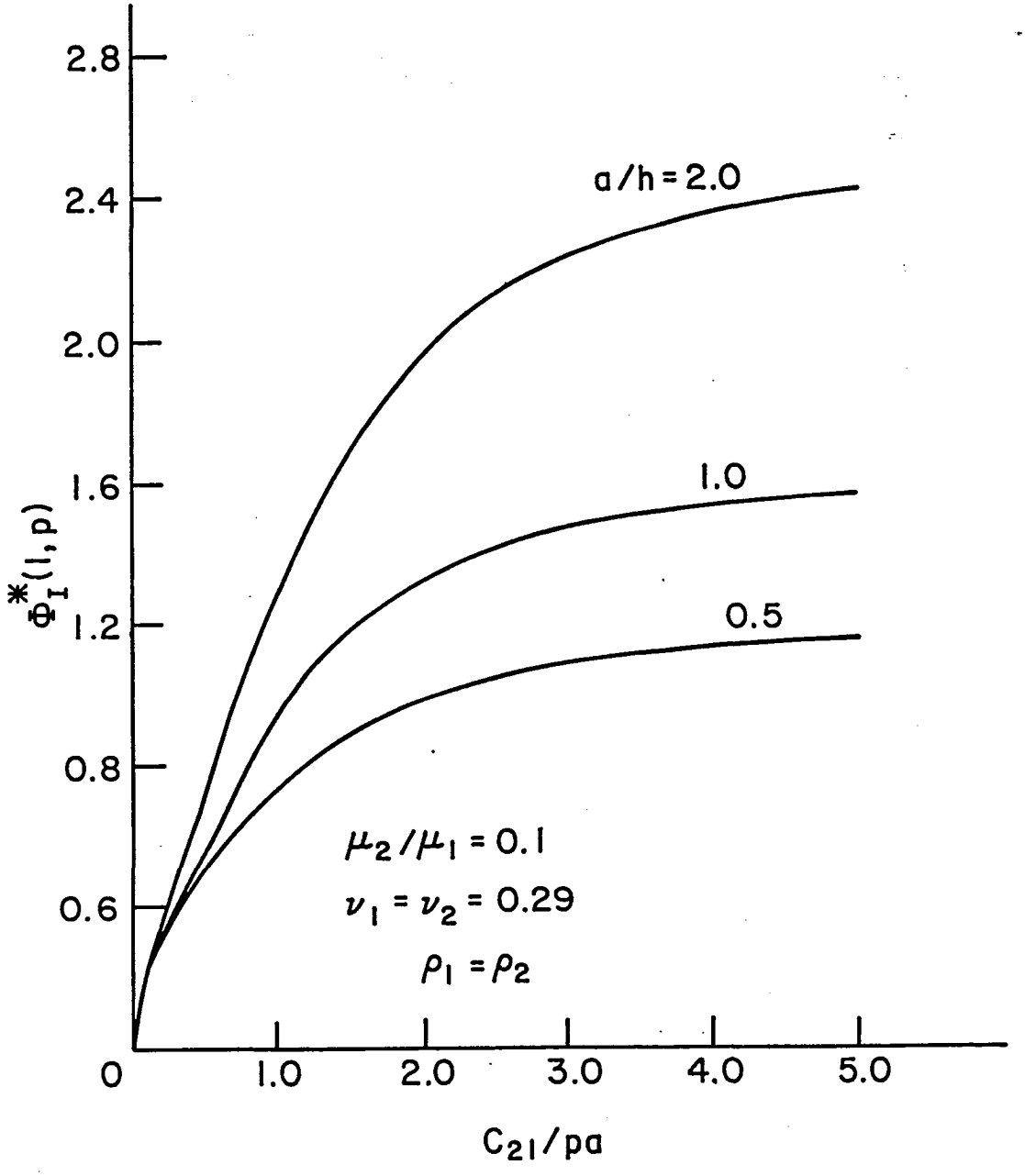

Figure 5. Variations of $\Phi_I^*(1,p)$ with c_{21}/pa for $\mu_2/\mu_1 = 0.1$

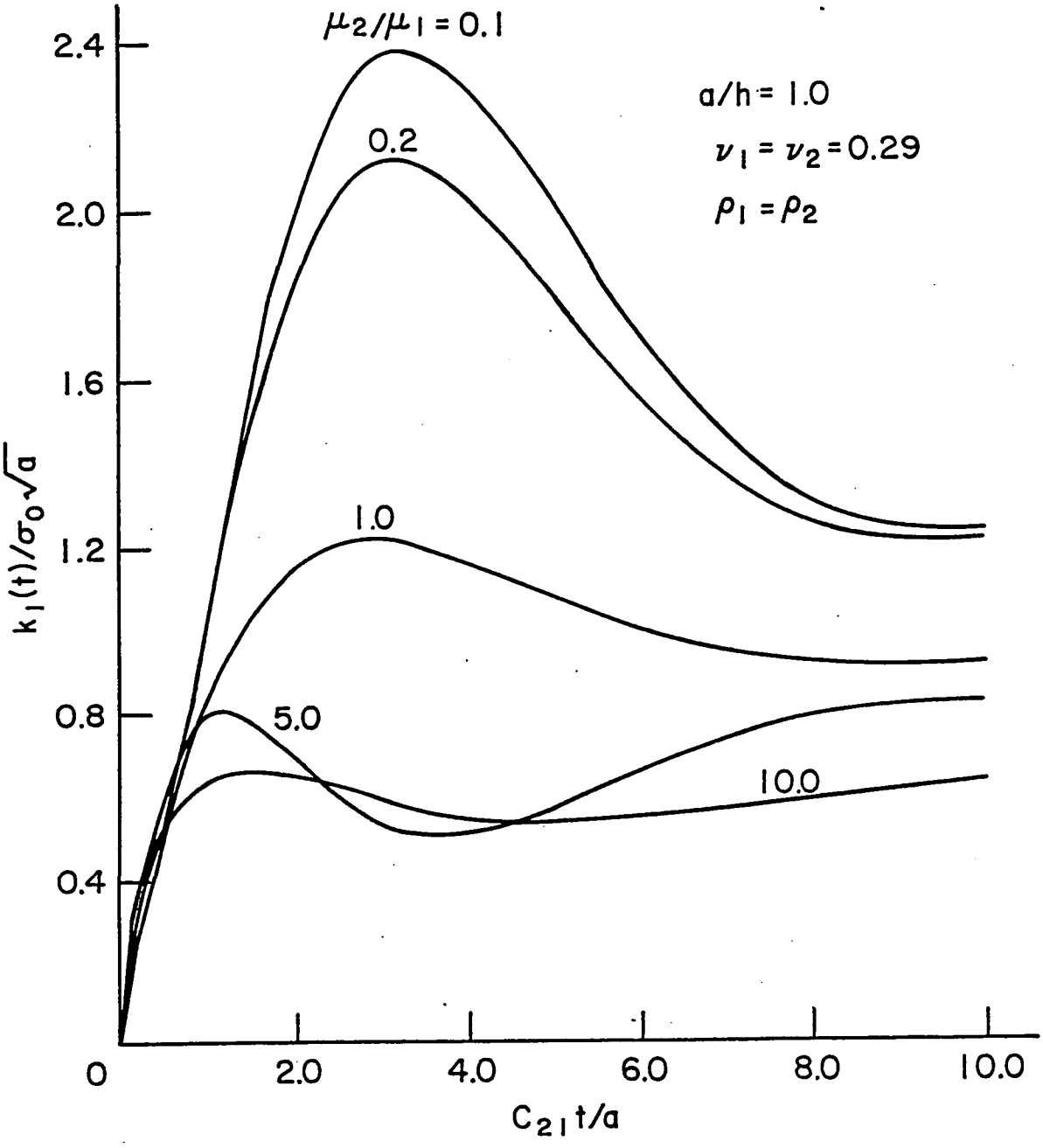

Figure 6. Dynamic stress intensity factor $k_1(t)$ versus time for a/h = 1.0

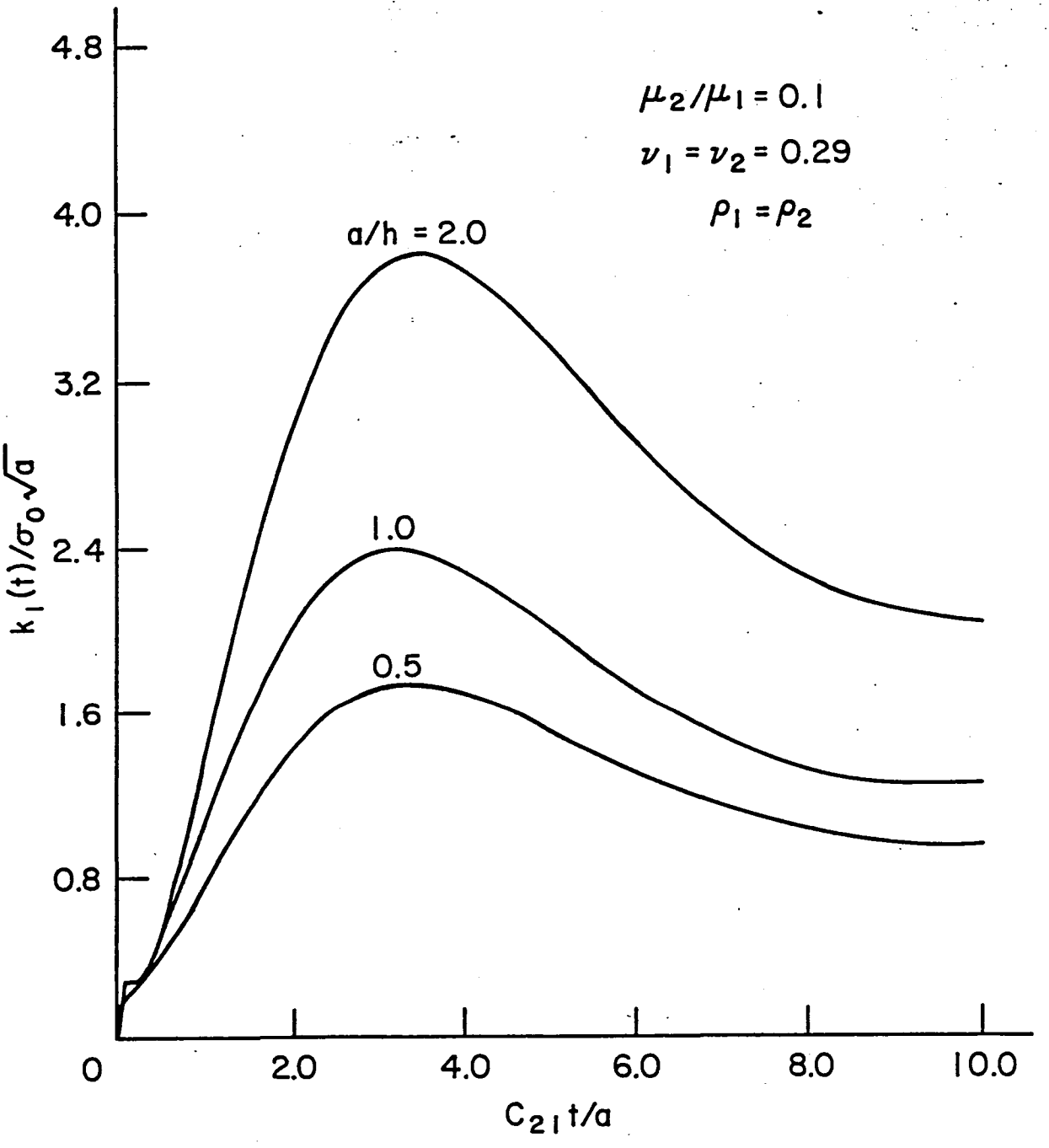

Figure 7. Dynamic stress intensity factor $k_1(t)$ versus time for $\mu_2/\mu_1 = 0.1$

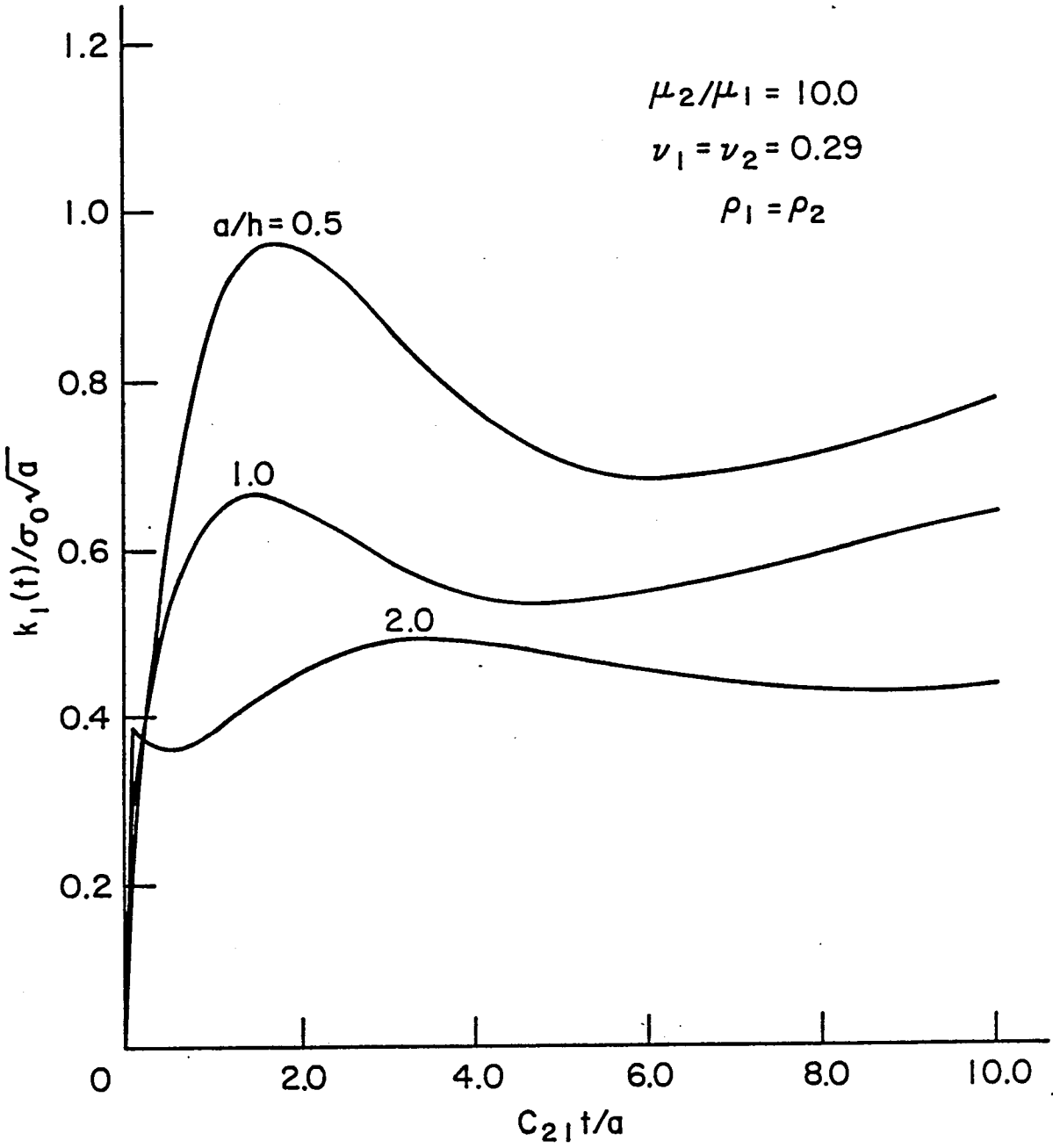

Figure 8. Dynamic stress intensity factor $k_1(t)$ versus time for $\mu_2/\mu_1 = 10.0$

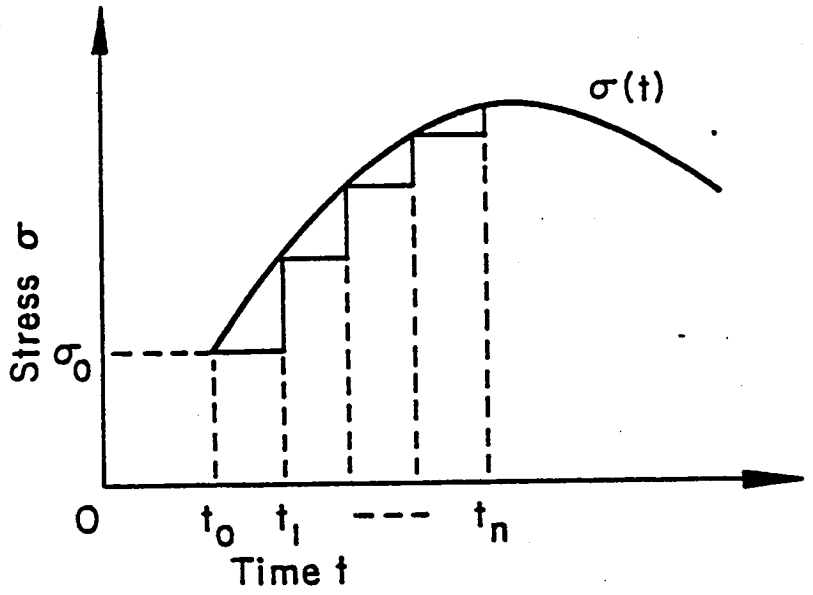

Figure 9. Applied stress as a general function of time

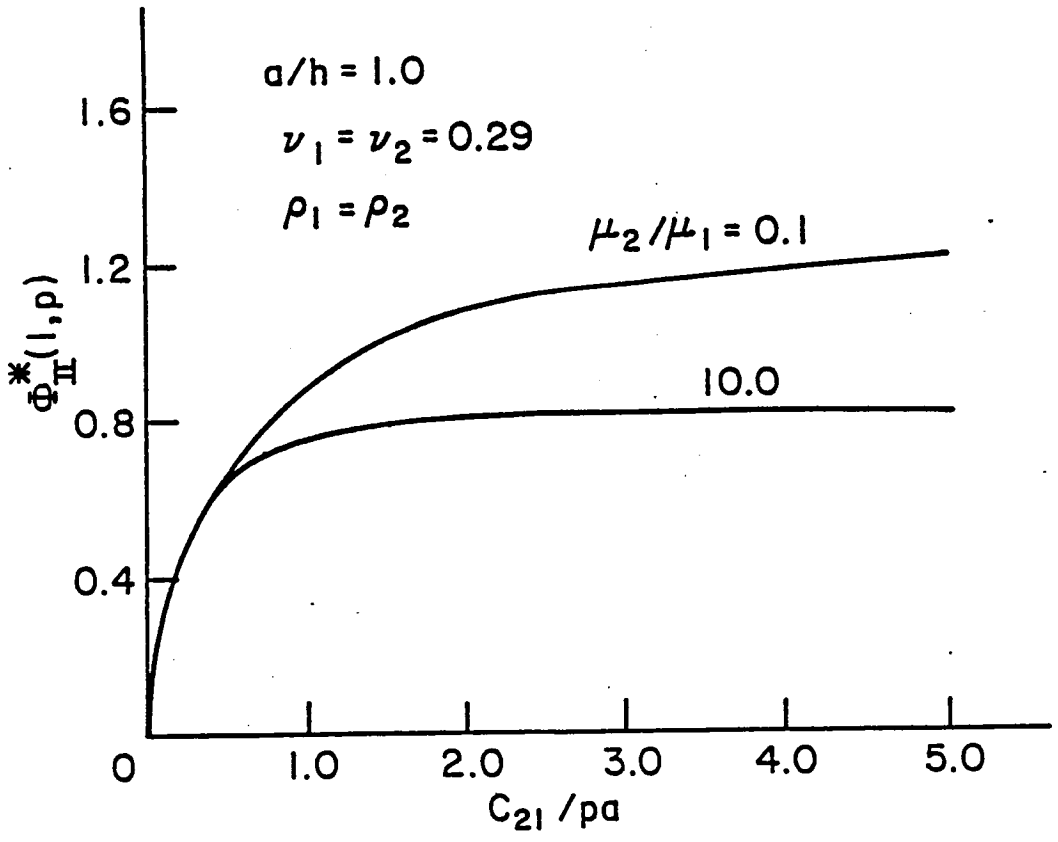

Figure 10. Variations of $\Phi^*_{II}(1,p)$ with c_{21}/pa

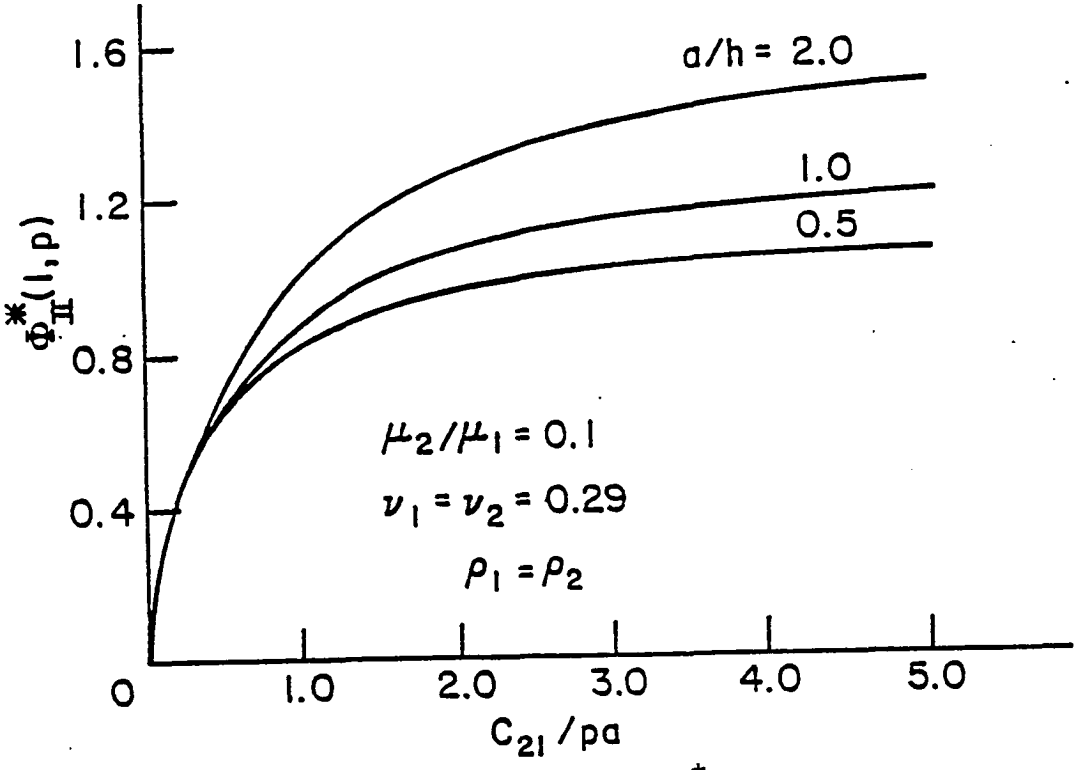

Figure 11. Variations of $\Phi^*_{II}(1,p)$ with c_{21}/pa

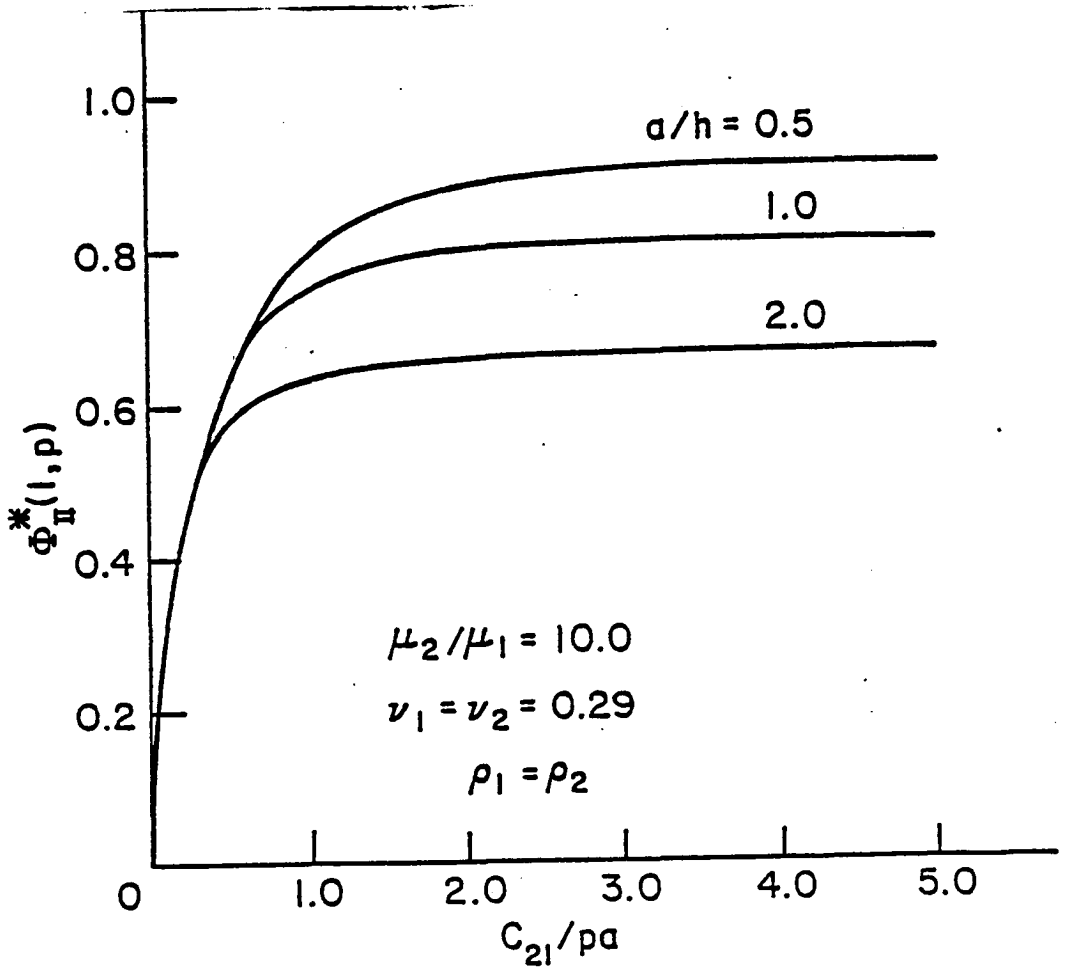

Figure 12. Variations of $\Phi^*_{II}(1,p)$ with c_{21}/pa

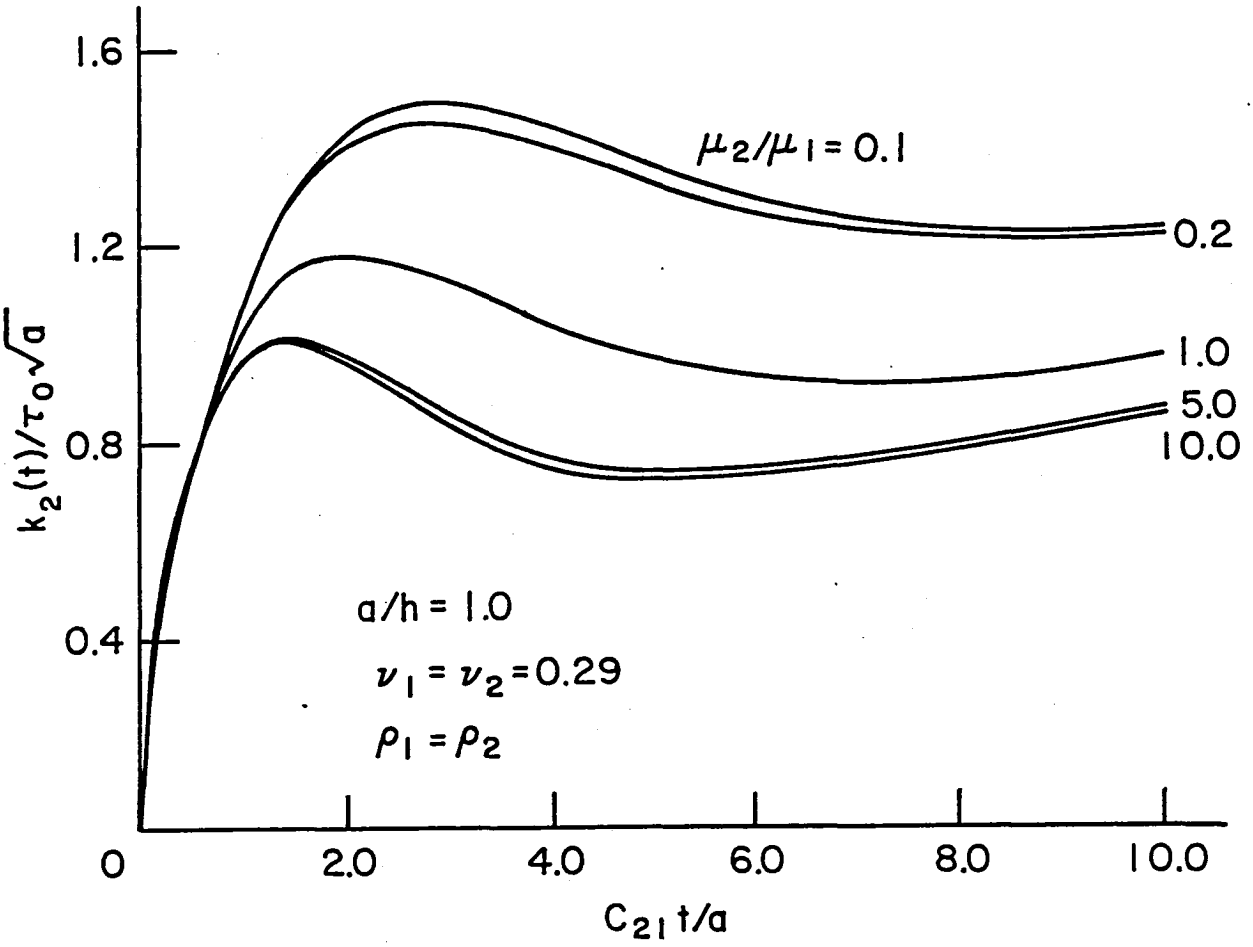

Figure 13. Dynamic stress intensity factor $k_2(t)$ versus time for a/h = 1.0

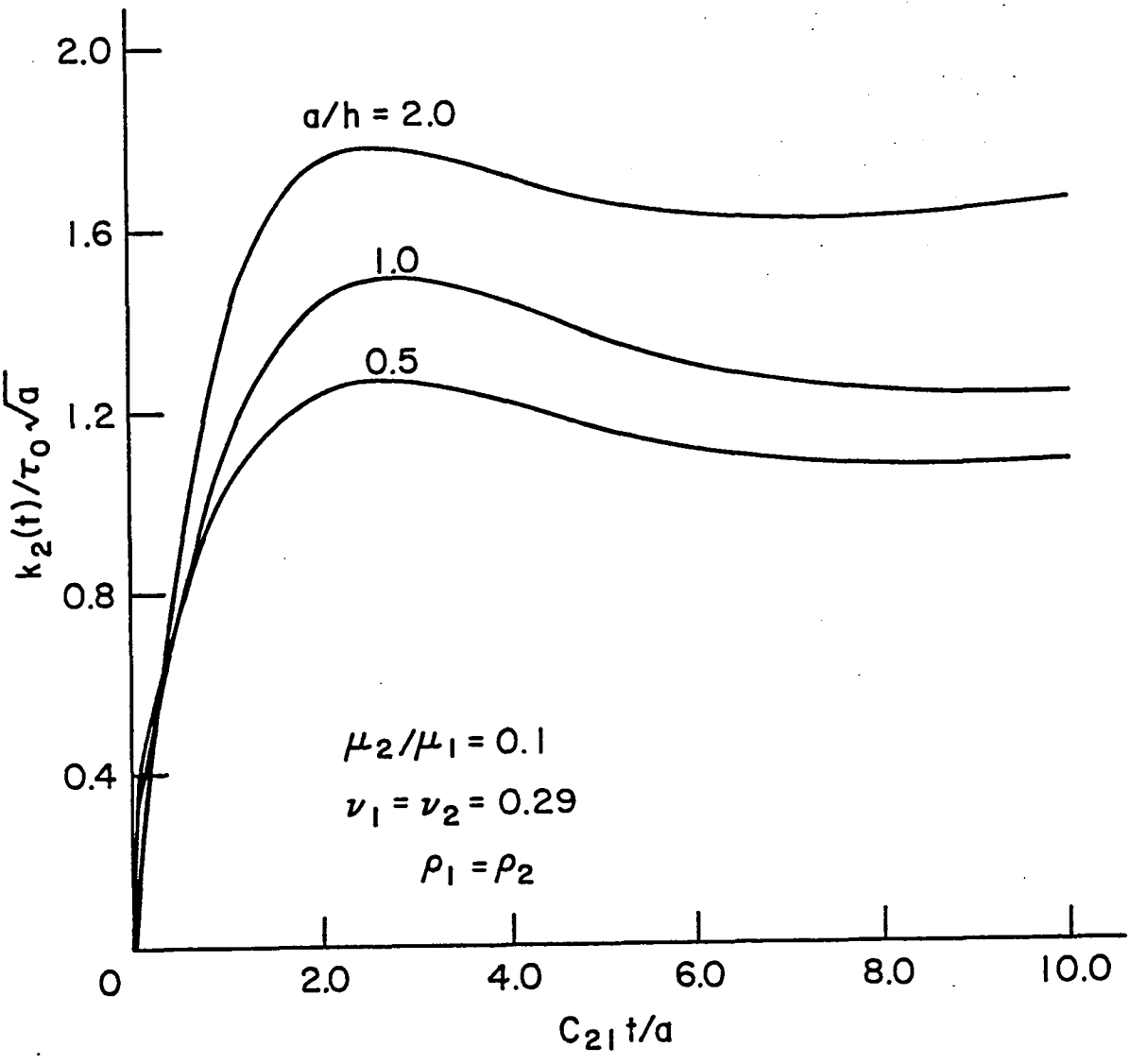

Figure 14. Dynamic stress intensity factor $k_2(t)$ versus time for $\mu_2/\mu_1 = 0.1$

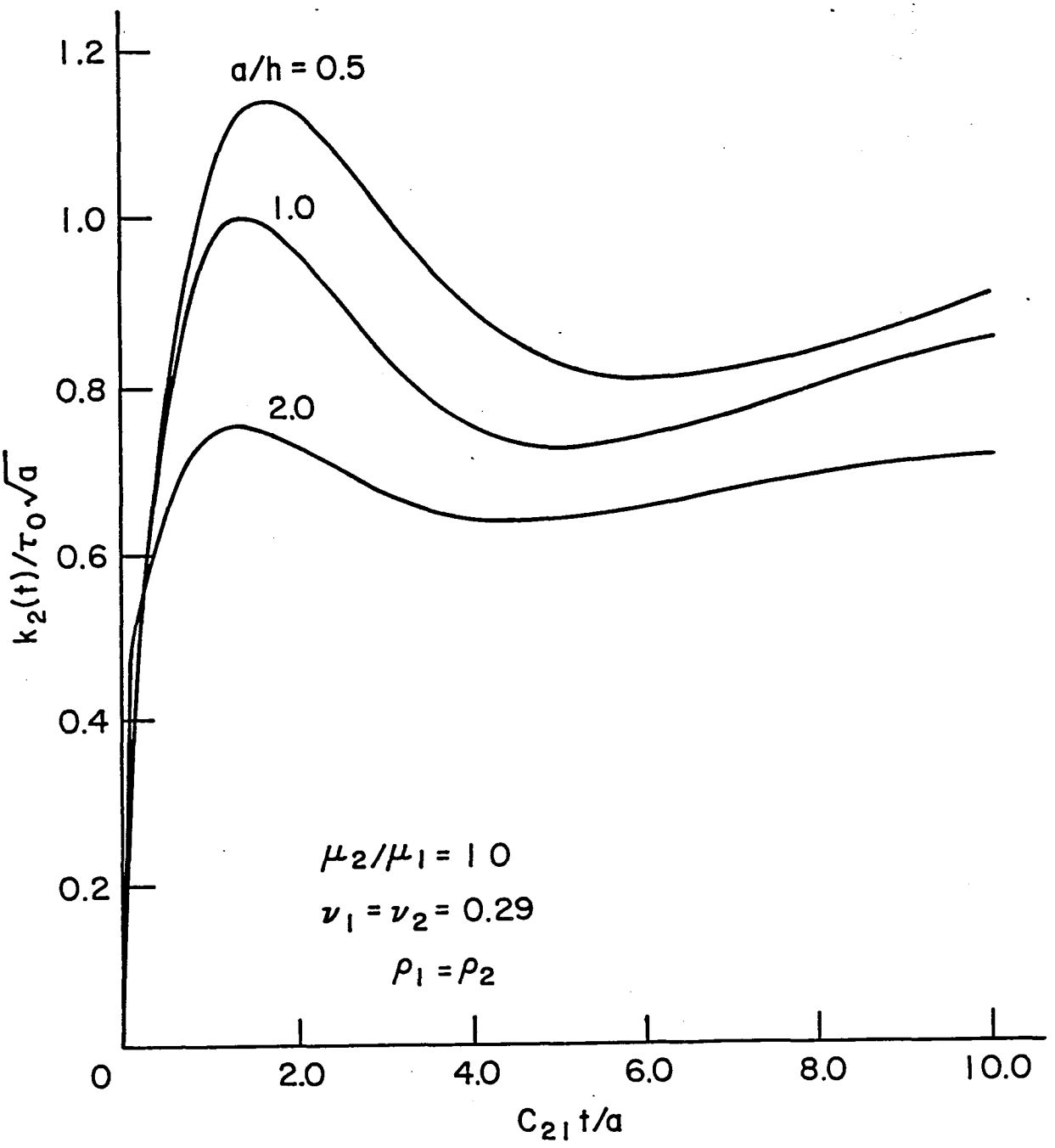

Figure 15. Dynamic stress intensity factor $k_2(t)$ versus time for $\mu_2/\mu_1 = 10.0$

Normal impact.

```
                PROGRAM BETA(INPUT,OUTPUT,PUNCH,PLOT,TAPE 99=PLOT)
          3     REAL NON(4),F(4,4,1),G(4,4),D(4),PT(4)
          3     REAL B(4),C(4)
          3     REAL LP(50),DTA(50)
          3     EQUIVALENCE (NON,B)
          3      COMMON K1,K2,K3,K4
          3     COMMON/AUX/H,P,PK1,PK2,BMU,X,Y
          3     LP(1)=0.0
          4     DTA(1)=0.0
          5     READ 2,K1,K2,K3,K4
         20   2  FORMAT(I2)
              *  K1 = ORDER OF SYSTEM OF EQUATIONS
              *  K2 = NO. OF DISTINCT KERNELS
              *  K3 = NO. OF DATA POINTS
              *  K4 = NO. OF DATA SETS TO BE EVALUATED
              *  SET UP DATA POINTS
         20       AK=K3
         22       DO 5 N=1,K3
         23       AN=N
         24   5   PT(N)=AN/AK
              *  SET UP INTEGRATION MATRIX
         31       M=K3-2
         33       N=K3-1
         34       A=K3
         35       A=1./(3.*A)
         37       DO 10 K=2,M,2
         41   10  D(K)=2.*A
         46       DO 15 K=1,N,2
         47   15  D(K)=4.*A
         54       D(K3)=A
              *  CALCULATE NONHOMOGENEOUS TERMS
         56       RHS=1.0
         57       DO 22 I=1,K2
         61       PRINT 9
         64    9  FORMAT(1H1)
         64       READ 61,BMU
         72   61  FORMAT(F10.5)
         72       DO 999 II=1,K4
         74       DO 35 N=1,K3
         75   35  NON(N)=RHS*SQRT(PT(N))
              *  CALCULATE KERNEL MATRICES
        104       CALL CONST(I)
        106       DO 20 N=1,K3
        110       DO 20 M=1,K3
        111       IF(M-N)25,30,30
        114   25  F(M,N,I)=F(N,M,I)
        123       GO TO 20
        123   30  F(M,N,I)=FU(I,PT(M),PT(N))
        134   20  CONTINUE
        141       CALL CHANGE(F,G,D,I)
        144       CALL LINEQ(G,B,C,   K3)
        147        DO 40 L=1,K3
        151       PRINT 6,PT(L),NON(L)
        160    6  FORMAT(5X,F8.4,F15.6)
        160   40  CONTINUE
        163       LP(II+1)=NON(K3)
        165       DTA(II+1)=P
        167  999  CONTINUE
        171       PUNCH 66,(DTA(IX),LP(IX),IX=1,19)
        205    66 FORMAT(2F10.5)
        205       CALL LAPINV(DTA,LP)
        207   22  CONTINUE
        212       END


                  FUNCTION SIMF(I,A,B)
                  COMMON/AUX/H,P,PK1,PK2,BMU,X,Y
          6       DEL=0.25*(B-A)
          6       IF(DEL)40,45,50
         10   45  SIMP=0.0
         12       RETURN
         13   50  CONTINUE
         14       SA=Z(I,A)+Z(I,B)
         14       SB=Z(I,A+2.*DEL)
         26       SC=Z(I,A+DEL)+Z(I,A+3.*DEL)
         35
```

```
 53            S1=(DEL/3.)*(SA+2.*SB+4.*SC)
 61            IF(S1.EQ.0.0) GO TO 45
 62            K=8
 63       35   SB=SB+SC
 65            DEL=0.5*DEL
 67            SC=Z(I,A+DEL)
 75            J=K-1
 77            DO 5 N=3,J,2
100            AN=N
101       5    SC=SC+Z(I,A+AN*DEL)
113            S2=(DEL/3.)*(SA+2.*SB+4.*SC)
122            DIF=ABS((S2-S1)/S1)
125            ER=0.01
127            IF(DIF-ER)30,25,25
131       30   SIMP=S2
133            RETURN
133       25   K=2*K
134            S1=S2
136            IF(K-2048)35,35,40
140       40   PRINT 42,I,A,B
152       42   FORMAT(5X,* INT. DOES NOT CONVERGE *,I3,2F9.4)
152            PRINT 60,X,Y
162       60   FORMAT(2F10.5)
162            DO 70 J=1,10
166            DIP=J
167            DIP=DIP/10.
171            W=Z(I,DIP)
175            PRINT 60,W
202       70   CONTINUE
206            CALL EXIT
207            END


               SUBROUTINE CHANGE(F,G,D,I)
 7             REAL F(4,4,1),G(4,4),D(4)
 7             COMMON K1,K2,K3,K4
 7             DO 10 N=1,K3
10             DO 10 M=1,K3
11             G(M,N)   =F(M,N,I)*D(N)
24        10   CONTINUE
30             DO 20 N=1,K3
31        20   G(N,N)=G(N,N)+1.0
40             RETURN
41             END


               SUBROUTINE LINEQ(A,B,T,N)
 7             REAL A(N,N),B(N),T(N)
 7             DO 5 I=2,N
10         5   A(I,1)=A(I,1)/A(1,1)
17             DO 10 K=2,N
20             M=K-1
22             DO 15 I=1,N
23        15   T(I)=A(I,K)
33             DO 20 J=1,M
34             A(J,K)=T(J)
41             J1=J+1
43             DO 20 I=J1,N
44             T(I)=T(I)-A(I,J)*A(J,K)
55        20   CONTINUE
61             A(K,K)=T(K)
65             IF(K.EQ.N) GO TO 10
66             M=K+1
70             DO 25 I=M,N
71        25   A(I,K)=T(I)/A(K,K)
105       10   CONTINUE
          *  BACK SUBSTITUTE
110            DO 30 I=1,N
111            T(I)=B(I)
114            M=I+1
116            IF(M.GT.N) GO TO 30
121            DO 30 J=M,N
122            B(J)=B(J)-A(J,I)*T(I)
132       30   CONTINUE
136            DO 35 I=1,N
```

```
137             K=N+1-I
141             B(K)=T(K)/A(K,K)
146             K1=K-1
150             IF(K1.EQ.0) GO TO 35
151             DO 35 J1=1,K1
152             J=K-J1
154             T(J)=T(J)-A(J,K)*B(K)
162      35     CONTINUE
167             RETURN
167             END
```

```
                FUNCTION BESJO(A)
  3             IF(A-3.)5,5,10
  5         5   B=A*A/9.
  7             W=1.-2.2499997*B
 12             Z=B*B
 13             W=W+1.2656208*Z
 15             Z=Z*B
 16             W=W-.3163866*Z
 20             Z=Z*B
 22             W=W+.0444479*Z
 24             Z=Z*B
 25             W=W-.0039444*Z
 27             Z=Z*B
 31             BESJO=W+.00021*Z
 34             RETURN
 34        10   B=3./A
 36             W=.79788456-.00000077*B
 41             V=A-.78539816-.04166397*B
 44             Z=B*B
 45             W=W-.0055274*Z
 47             V=V-.00003954*Z
 52             Z=Z*B
 53             W=W-.00009512*Z
 55             V=V+.00262573*Z
 57             Z=Z*B
 61             W=W+.00137237*Z
 63             V=V-.00054125*Z
 65             Z=Z*B
 67             W=W-.00072805*Z
 71             V=V-.00029333*Z
 73             Z=Z*B
 75             W=W+.00014476*Z
 77             V=V+.00013558*Z
101             BESJO=W/SQRT(A)*COS(V)
111             RETURN
112             END
```

```
                FUNCTION FU(I,A,B)
  6             COMMON/AUX/H,P,PK1,PK2,BMU,X,Y
  6             X=A
  7             Y=B
 10             IF(A*B)5,10,5
 11        10   FU=0.0
 12             RETURN
 13         5   SUM=SIMF(I,0.0,5.0)
 20             ER=0.01
 21             DEL =5.0
 23        20   UP=DEL+5.0
 25             ADDL=SIMF(I,DEL,UP)
 32             DEL =UP
 33             TEST=ABS(ADDL/SUM)
 36             SUM=SUM+ADDL
 37             IF(TEST-ER)15,20,20
 41        15   FU=SQRT(X*Y)*SUM
 47             RETURN
 47             END
```

```
                SUBROUTINE CONST(I)
    3           COMMON/AUX/H,P,PK1,PK2,BMU,X,Y
    3           PR1=0.29
    5           PR2=0.29
    6           PK1=SQRT((1.-2.*PR1)/(2.*(1.-PR1)))
   15           PK2=SQRT((1.-2.*PR2)/(2.*(1.-PR2)))
   24           READ 1,P
   31         1 FORMAT(F10.5)
   31           HH=0.1
   33           HH=10.0
   34           HH=5.0
   36           HH=4.0
   37           HH=1.0
   41           HH=0.5
   42           HH=2.0
   44           H=1./HH
   45           PRINT 2,BMU,PR1,PR2,HH,P
   62         2 FORMAT(/////5X,* MU2/MU1 =*F6.2,* NU1 =*F4.2,* NU2 =*F4.2///5X,*
                1/H =*F4.2,* C21/PA =*F4.2)
   62           RETURN
   63           END


                FUNCTION Z(I,S)
    5           COMMON/AUX/H,P,PK1,PK2,BMU,X,Y
    5           PP=P*P
    6           C1=PK1*PK1
   10           C2=PK2*PK2
   11           CC=1.-C1
   13           GA=SQRT(S*S+C1/PP)
   22           GB=SQRT(S*S+1./PP)
   31           GC=SQRT(S*S+C2/BMU/PP)
   40           GD=SQRT(S*S+1./BMU/PP)
   47           AA=S*S+1./PP/2.
   53           AB=1.-BMU
   54           AC=S*S-GC*GD
   57           AD=(GB-GD)/AC/PP/2.*BMU
   64           AE=(GB+GD)/AC/PP/2.*BMU
   71           AF=(S*S-GA*GD)/AC/PP/2.*BMU
   77           AG=(S*S+GA*GD)/AC/PP/2.*BMU
  105           AH=(S*S-GB*GC)/AC/PP/2.*BMU
  113           AI=(S*S+GB*GC)/AC/PP/2.*BMU
  121           AJ=(GA-GC)/AC/PP/2.*BMU
  126           AK=(GA+GC)/AC/PP/2.*BMU
  133           A1=-(AB*GB-AD)
  136           A2=AB*GB-AE
  140           A3=AA-BMU*S*S-AF
  145           A4=AA-BMU*S*S-AG
  150           A5=-AA+BMU*S*S+AH
  154           A6=-AA+BMU*S*S+AI
  157           A7=S*(AB*GA-AJ)
  162           A8=-S*(AB*GA-AK)
  165           BA=A1*A6-A2*A5
  170           BB=A3*A6-S*A2*A7
  173           BC=A4*A6-S*A2*A8
  176           BD=S*A1*A7-A3*A5
  202           BE=S*A1*A8-A4*A5
  205           B1=BB/BA
  207           B2=BC/BA
  211           B3=BD/BA
  212           B4=BE/BA
  214           EA=2.*GA*H
  216           EB=2.*GB*H
  220           EC=(EA+EB)/2.
  222           ED=2.*EC
  223           E1=EXP(-EA)
  226           E2=EXP(-EB)
  232           E3=EXP(-EC)
  236           E4=EXP(-ED)
  242           DL=B2+B3*E4+B4*E2+B1*E1
  251           D1=2.*PP/CC/GA/DL
  255           D2=AA*AA-S*S*GA*GB
  263           D3=B2-B3*E4
  266           D4=2.*AA*(GB*(B1*B4-B2*B3)-S*S*GA)*E3
  275           D5=(AA*AA+S*S*GA*GB)*(B4*E2-B1*E1)
```

```
306           F=01*(02*03+04+05)
317           Z=(F-S)*BESJO(S*X)*BESJO(S*Y)
331           RETURN
330           END


              SUBROUTINE LAPINV(GLAM,PHI)
        C     THIS PROGRAM EVALUATES THE COEFFICIENTS FOR SERIES
        C     OF JACOBI POLYNOMIALS WHICH REPRESENTS A LAPLACE
        C     INVERSION INTEGRAL
              REAL MUL
  5           DIMENSION A(50),GLAM(50),PHI(50),C(4,50)
  5           DIMENSION BK(101),TT(101)
  5           COMMON/2/TI,TF,DT,MN,BK,TT
  5           READ 1,NN,MN,MM
 16         1 FORMAT(3I2)
 16           READ 2,TI,TF,DT
 30         2 FORMAT(3F10.5)
 30           PRINT 99
 34        99 FORMAT(1H1)
 34           CALL SPLICE(GLAM,PHI,MM,C)
 40           PRINT 101
 44       101 FORMAT(/////5X,*   GLAM              PHI     *)
 44           PRINT 102,(GLAM(I),PHI(I),I=1,MM)
 65       102 FORMAT(5X,F10.5,5X,F10.5)
 65           M11=MM-1
 67           PRINT 300
 73       300 FORMAT(////5X,*   C(1,J)          C(2,J)          C(3,J)          C(4
             1,J)  *)
 73           PRINT 103,((C(I,J),I=1,4),J=1,M11)
112       103 FORMAT(5X,F10.5,5X,F10.5,5X,F10.5,5X,F10.5)
112           PRINT 99
116           DO 10 I=1,NN
121           READ 3,BET,DEL
130         3 FORMAT(2F10.5)
130           PRINT 98,BET,DEL
140        98 FORMAT(/////5X,*BETA =*F5.3,* DELTA =*F5.3)
140           DO 11 L=1,MN
143           AL=L
144           S=1./(AL+BET)/DEL
150           CALL SPLINE(GLAM,PHI,MM,C,S,G)
153           F=G*S
155           IF(AL-2.)81,82,83
161        81 A(1)=(1.+BET)*DEL*F
165           GO TO 11
165        82 A(2)=((2.+BET)*DEL*F-A(1))*(3.+BET)
175           GO TO 11
175        83 CONTINUE
175           TOP=1.
177           L1=L-1
201           AL1=L1
202           DO 12 J=1,L1
203           AJ=J
204           TOP=AJ*TOP
206        12 CONTINUE
210           L2=2*L-1
212           BOT=1.
214           DO 13 J=L,L2
215           AJ=J
216           BOT=(AJ+BET)*BOT
221        13 CONTINUE
223           MUL=BOT/TOP
225           SUM=0.0
226           DO 14 N=1,L1
227           AN=N
230           IF(AN-2.)85,86,87
233        85 TOD=1.
235           GO TO 88
235        86 TOD=AL1
237           GO TO 88
237        87 CONTINUE
237           TOD=1.
241           ICH=L1-(N-2)
244           DO 15 J=ICH,L1
245           AJ=J
246           TOD=AJ*TOD
```

```
250         15 CONTINUE
252         88 CONTINUE
252            BOD=1.
254            JA=L1+N
256            DO 16 J=L,JA
260            AJ=J
261            BOD=BOD*(AJ+BET)
264         16 CONTINUE
266            CO=TOD/BOD
270            SUM=SUM+CO*A(N)
273         14 CONTINUE
275            A(L)=MUL*(DEL*F-SUM)
301         11 CONTINUE
304            CALL JACSER(DEL,A,BET)
306            CALL NAMPLT
307            CALL QIKSET(6.0,0.0,0.0,6.0,0.0,0.0)
313            CALL QIKSAX(3,3)
315            CALL QIKPLT(TT,BK,101)
320            CALL ENDPLT
321         10 CONTINUE
325        999 CONTINUE
325            RETURN
326            END


               SUBROUTINE JACSER(D,C,B)
  6            DIMENSION C(50),SF(50),P(50)
  6            DIMENSION BK(101),TT(101)
  6            COMMON/2/TI,TF,DT,MN,BK,TT
  6            TT(1)=0.0
  7            BK(1)=0.0
 10            LM=1
 11            T=TI
 12         12 T=T+DT
 14            X=2.*EXP(-D*T)-1.
 24            CALL JACOBI(MN,X,B,P)
 26            SF(1)=C(1)*P(1)
 32            DO 10 L=2,MN
 33            L1=L-1
 35            AL=L
 36            SF(L)=SF(L1)+C(L)*P(L)
 43         10 CONTINUE
 45            PRINT 97,T,X
 55         97 FORMAT(/////5X,*  T =*F6.3,*  X =*F10.5)
 55            PRINT 96
 61         96 FORMAT(///5X,* I   C(I)   *,5X,* N   F(T)   *)
 61            DO 11 I=1,6
 65            PRINT 95,I,C(I),I,SF(I)
105         95 FORMAT(5X,I2,F10.2,5X,I2,F10.5)
105         11 CONTINUE
111            LM=LM+1
113            BK(LM)=SF(5)
115            TT(LM)=T
117            IF(T.LE.TF) GO TO 12
121            RETURN
122            END


               SUBROUTINE JACOBI(N,X,B,PB)
     C         THIS PROGRAM CALCULATES JACOBI POLYNOMIALS OF ORDER
     C         K-1 WITH ARG X AND PARAMETER B GT -1
  7            DIMENSION PB(N)
  7            AN=N
 10            IF(AN-2.)1,2,3
 12          1 PB(1)=1.
 14            RETURN
 14          2 PB(1)=1.
 16            PB(2)=X-B*(1.-X)/2.
 21            RETURN
 22          3 BSQ=B*B
 23            BONE=B+1.
 25            PB(1)=1.
 26            PB(2)=X-B*(1.-X)/2.
 31            DO 4 K=3,N
 33            AK=K
```

```
 34           AK1=AK-1.
 36           AK2=AK-2.
 40           K1=K-1
 42           K2=K-2
 43           CO1=((2.*AK1)+B)*X
 46           CO1=((2.*AK2)+B)*CO1
 51           CO1=((2.*AK2)+BONE)*(CO1-BSQ)
 56           CO2=2.*AK2*(AK2+B)*((2.*AK1)+B)
 64           CO=2.*AK1*(AK1+B)*((2.*AK2)+B)
 71         4 PB(K)=(CO1*PB(K1)-CO2*PB(K2))/CO
102           RETURN
103           END


              SUBROUTINE SPLINE(X,Y,M,C,XINT,YINT)
 11           DIMENSION X(50),Y(50),C(4,50)
 11           IF(XINT-X(1))1,10,11
 13        10 YINT=Y(1)
 14           RETURN
 15        11 CONTINUE
 15           IF(X(M)-XINT)1,12,13
 21        12 YINT=Y(M)
 23           RETURN
 23        13 CONTINUE
 23           K=M/2
 25           N=M
 26         2 CONTINUE
 26           IF(X(K)-XINT)3,14,5
 32        14 YINT=Y(K)
 34           RETURN
 35         3 CONTINUE
 35           IF(XINT-X(K+1))4,15,7
 41        15 YINT=Y(K+1)
 43           RETURN
 43         4 CONTINUE
 43           YINT=(X(K+1)-XINT)*(C(1,K)*(X(K+1)-XINT)**2+C(3,K))
 54           YINT=YINT+(XINT-X(K))*(C(2,K)*(XINT-X(K))**2+C(4,K))
 65           RETURN
 65         5 CONTINUE
 65           IF(X(K-1)-XINT)6,16,17
 70         6 K=K-1
 72           GO TO 4
 72        16 YINT=Y(K-1)
 74           RETURN
 75        17 N=K
 77           K=K/2
100           GO TO 2
100         7 LL=K
102           K=(N+K)/2
103         8 CONTINUE
103           IF(X(K)-XINT)3,14,18
106        18 CONTINUE
106           IF(X(K-1)-XINT)6,16,19
111        19 N=K
113           K=(LL+K)/2
114           GO TO 8
115         1 PRINT 101
121       101 FORMAT(* OUT OF RANGE FOR INTERPOLATION  *)
121           STOP
123           END


              SUBROUTINE SPLICE(X,Y,M,C)
  7           DIMENSION X(50),Y(50),D(50),P(50),E(50),C(4,50)
  7           DIMENSION A(50,3),B(50),Z(50)
  7           MM=M-1
 11           DO 2 K=1,MM
 12           D(K)=X(K+1)-X(K)
 15           P(K)=D(K)/6.
 20         2 E(K)=(Y(K+1)-Y(K))/D(K)
 26           DO 3 K=2,MM
 27         3 B(K)=E(K)-E(K-1)
 34           A(1,2)=-1.-D(1)/D(2)
 37           A(1,3)=D(1)/D(2)
 41           A(2,3)=P(2)-P(1)*A(1,3)
```

```
      A(2,2)=2.*(P(1)+P(2))-P(1)*A(1,2)
      A(2,3)=A(2,3)/A(2,2)
      B(2)=B(2)/A(2,2)
      DO 4 K=3,MM
      A(K,2)=2.*(P(K-1)+P(K))-P(K-1)*A(K-1,3)
      B(K)=B(K)-P(K-1)*B(K-1)
      A(K,3)=P(K)/A(K,2)
    4 B(K)=B(K)/A(K,2)
      Q=D(M-2)/D(M-1)
      A(M,1)=1.+Q+A(M-2,3)
      A(M,2)=-Q-A(M,1)*A(M-1,3)
      B(M)=B(M-2)-A(M,1)*B(M-1)
      Z(M)=B(M)/A(M,2)
      MN=M-2
      DO 6 I=1,MN
      K=M-I
    6 Z(K)=B(K)-A(K,3)*Z(K+1)
      Z(1)=-A(1,2)*Z(2)-A(1,3)*Z(3)
      DO 7 K=1,MM
      Q=1./(6.*D(K))
      C(1,K)=Z(K)*Q
      C(2,K)=Z(K+1)*Q
      C(3,K)=Y(K)/D(K)-Z(K)*P(K)
    7 C(4,K)=Y(K+1)/D(K)-Z(K+1)*P(K)
      RETURN
      END
```

```
                      PROGRAM BETA(INPUT,OUTPUT,PUNCH,PLOT,TAPE 99=PLOT)
    3                 REAL NON(4),F(4,4,1),G(4,4),D(4),PT(4)
    3                 REAL B(4),C(4)
    3                 REAL LP(50),DTA(50)
    3                 EQUIVALENCE (NON,B)
    3                  COMMON K1,K2,K3,K4
    3                 COMMON/AUX/H,P,PK1,PK2,BMU,X,Y
    3                 LP(1)=0.0
    4                 DTA(1)=0.0
    5                 READ 2,K1,K2,K3,K4
   20       2         FORMAT(I2)
            *   K1 = ORDER OF SYSTEM OF EQUATIONS
            *   K2 = NO. OF DISTINCT KERNELS
            *   K3 = NO. OF DATA POINTS
            *   K4 = NO. OF DATA SETS TO BE EVALUATED
            *   SET UP DATA POINTS
   21                 AK=K3
   22                 DO 5 N=1,K3
   23                 AN=N
   24       5         PT(N)=AN/AK
            *   SET UP INTEGRATION MATRIX
   31                 M=K3-2
   33                 N=K3-1
   34                 A=K3
   35                 A=1./(3.*A)
   37                 DO 10 K=2,M,2
   41      10         D(K)=2.*A
   46                 DO 15 K=1,N,2
   47      15         D(K)=4.*A
   54                 D(K3)=A
            *   CALCULATE NONHOMOGENEOUS TERMS
   56                 RHS=1.0
   57                 DO 22 I=1,K2
   61                 PRINT 9
   64       9         FORMAT(1H1)
   64                 READ 61,BMU
   72      61         FORMAT(F10.5)
   72                 DO 999 II=1,K4
   74                 DO 35 N=1,K3
   75      35         NON(N)=RHS*SQRT(PT(N))
            *   CALCULATE KERNEL MATRICES
  104                 CALL CONST(I)
  106                 DO 20 N=1,K3
  110                 DO 20 M=1,K3
  111                 IF(M-N)25,30,30
  114      25         F(M,N,I)=F(N,M,I)
  123                 GO TO 20
  123      30         F(M,N,I)=FU(I,PT(M),PT(N))
  134      20         CONTINUE
  141                 CALL CHANGE(F,G,D,I)
  144                 CALL LINEQ(G,B,C,   K3)
  147                  DO 40 L=1,K3
  151                 PRINT 6,PT(L),NON(L)
  160       6         FORMAT(5X,F8.4,F15.6)
  160      40         CONTINUE
  163                 LP(II+1)=NON(K3)
  165                 DTA(II+1)=P
  167     999         CONTINUE
  171                 PUNCH 66,(DTA(IX),LP(IX),IX=1,19)
  205      66         FORMAT(2F10.5)
  205                 CALL LAPINV(DTA,LP)
  207      22         CONTINUE
  212                 END



                      FUNCTION SIMP(I,A,B)
    6                 COMMON/AUX/H,P,PK1,PK2,BMU,X,Y
    6                 DEL=0.25*(B-A)
   10                 IF(DEL)40,45,50
   12      45         SIMP=0.0
   13                 RETURN
   14      50         CONTINUE
   14                 SA=Z(I,A)+Z(I,B)
   26                 SB=Z(I,A+2.*DEL)
   35                 SC=Z(I,A+DEL)+Z(I,A+3.*DEL)
```

```
 53            S1=(DEL/3.)*(SA+2.*SB+4.*SC)
 61            IF(S1.EQ.0.0) GO TO 45
 62            K=8
 63       35   SB=SB+SC
 65            DEL=0.5*DEL
 67            SC=Z(I,A+DEL)
 75            J=K-1
 77            DO 5 N=3,J,2
100            AN=N
101       5    SC=SC+Z(I,A+AN*DEL)
113            S2=(DEL/3.)*(SA+2.*SB+4.*SC)
122            DIF=ABS((S2-S1)/S1)
125            ER=0.01
127            IF(DIF-ER)30,25,25
131       30   SIMP=S2
133            RETURN
133       25   K=2*K
134            S1=S2
136            IF(K-2048)35,35,40
140       40   PRINT 42,I,A,B
152        42  FORMAT(5X,* INT. DOES NOT CONVERGE *,I3,2F9.4)
152            PRINT 60,X,Y
162       60   FORMAT(2F10.5)
162            DO 70 J=1,10
166            DIP=J
167            DIP=DIP/10.
171            W=Z(I,DIP)
175            PRINT 60,W
202       70   CONTINUE
206            CALL EXIT
207            END


               SUBROUTINE CHANGE(F,G,D,I)
  7            REAL F(4,4,1),G(4,4),D(4)
  7            COMMON K1,K2,K3,K4
  7            DO 10 N=1,K3
 10            DO 10 M=1,K3
 11            G(M,N)  =F(M,N,I)*D(N)
 24       10   CONTINUE
 30            DO 20 N=1,K3
 31       20   G(N,N)=G(N,N)+1.0
 40            RETURN
 41            END


               SUBROUTINE LINEQ(A,B,T,N)
  7            REAL A(N,N),B(N),T(N)
  7            DO 5 I=2,N
 10        5   A(I,1)=A(I,1)/A(1,1)
 17            DO 10 K=2,N
 20            M=K-1
 22            DO 15 I=1,N
 23       15   T(I)=A(I,K)
 33            DO 20 J=1,M
 34            A(J,K)=T(J)
 41            J1=J+1
 43            DO 20 I=J1,N
 44            T(I)=T(I)-A(I,J)*A(J,K)
 55       20   CONTINUE
 61            A(K,K)=T(K)
 65            IF(K.EQ.N) GO TO 10
 66            M=K+1
 70            DO 25 I=M,N
 71       25   A(I,K)=T(I)/A(K,K)
105       10   CONTINUE
          *  BACK SUBSTITUTE
110            DO 30 I=1,N
111            T(I)=B(I)
114            M=I+1
116            IF(M.GT.N) GO TO 30
121            DO 30 J=M,N
122            B(J)=B(J)-A(J,I)*T(I)
132       30   CONTINUE
136            DO 35 I=1,N
```

```
137             K=N+1-I
141             B(K)=T(K)/A(K,K)
146             K1=K-1
150             IF(K1.EQ.0) GO TO 35
151             DO 35 J1=1,K1
152             J=K-J1
154             T(J)=T(J)-A(J,K)*B(K)
162       35    CONTINUE
167             RETURN
167             END


                FUNCTION BESJO(A)
  3             IF(A-3.)5,5,10
  5        5    B=A*A/9.
  7             W=1.-2.2499997*B
 12             Z=B*B
 13             W=W+1.2656208*Z
 15             Z=Z*B
 16             W=W-.3163866*Z
 20             Z=Z*B
 22             W=W+.0444479*Z
 24             Z=Z*B
 25             W=W-.0039444*Z
 27             Z=Z*B
 31             BESJO=W+.0021*Z
 34             RETURN
 34       10    B=3./A
 36             W=.79786456-.00000077*B
 41             V=A-.78539816-.04166397*B
 44             Z=B*B
 45             W=W-.0055274*Z
 47             V=V-.0000954*Z
 52             Z=Z*B
 53             W=W-.0009512*Z
 55             V=V+.0026573*Z
 57             Z=Z*B
 61             W=W+.0013723*Z
 63             V=V-.0010425*Z
 65             Z=Z*B
 67             W=W-.0007280*Z
 71             V=V-.0002933*Z
 73             Z=Z*B
 75             W=W+.0001476*Z
 77             V=V+.0001358*Z
101             BESJO=W/SQRT(A)*COS(V)
111             RETURN
112             END


                FUNCTION FU(I,A,B)
  6             COMMON/AUX/H,P,PK1,PK2,BMU,X,Y
  6             X=A
  7             Y=B
 10             IF(A*B)5,10,5
 11       10    FU=0.0
 12             RETURN
 13        5    SUM=SIMP(I,0.0,5.0)
 20             ER=0.01
 21             DEL =5.0
 23       20    UP=DEL+5.0
 25             ADDL=SIMP(I,DEL,UP)
 32             DEL =UP
 33             TEST=ABS(ADDL/SUM)
 36             SUM=SUM+ADDL
 37             IF(TEST-ER)15,20,20
 41       15    FU=SQRT(X*Y)*SUM
 47             RETURN
 47             END
```

```
                SUBROUTINE CONST(I)
  3             COMMON/AUX/H,P,PK1,PK2,BMU,X,Y
  3             PR1=0.29
  5             PR2=0.29
  6             PK1=SQRT((1.-2.*PR1)/(2.*(1.-PR1)))
 15             PK2=SQRT((1.-2.*PR2)/(2.*(1.-PR2)))
 24             READ 1,P
 31           1 FORMAT(F10.5)
 31             HH=0.1
 33             HH=10.0
 34             HH=5.0
 36             HH=4.0
 37             HH=0.5
 41             HH=1.0
 42             HH=2.0
 44             H=1./HH
 45             PRINT 2,BMU,PR1,PR2,HH,P
 62           2 FORMAT(/////5X,* MU2/MU1 =*F6.2,* NU1 =*F4.2,* NU2 =*F4.2///5X,* A
               1/H =*F4.2,* C21/PA =*F4.2)
 62             RETURN
 63             END



                FUNCTION Z(I,S)
  5             COMMON/AUX/H,P,PK1,PK2,BMU,X,Y
  5             PP=P*P
  6             C1=PK1*PK1
 10             C2=PK2*PK2
 11             CC=1.-C1
 13             GA=SQRT(S*S+C1/PP)
 22             GB=SQRT(S*S+1./PP)
 31             GC=SQRT(S*S+C2/BMU/PP)
 40             GD=SQRT(S*S+1./BMU/PP)
 47             AA=S*S+1./PP/2.
 53             AB=1.-BMU
 54             AC=S*S-GC*GD
 57             AD=(GB-GD)/AC/PP/2.*BMU
 64             AE=(GB+GD)/AC/PP/2.*BMU
 71             AF=(S*S-GA*GD)/AC/PP/2.*BMU
 77             AG=(S*S+GA*GD)/AC/PP/2.*BMU
105             AH=(S*S-GB*GC)/AC/PP/2.*BMU
113             AI=(S*S+GB*GC)/AC/PP/2.*BMU
121             AJ=(GA-GC)/AC/PP/2.*BMU
126             AK=(GA+GC)/AC/PP/2.*BMU
133             A1=-(AB*GB-AD)
136             A2=AB*GB-AE
140             A3=AA-BMU*S*S-AF
145             A4=AA-BMU*S*S-AG
150             A5=-AA+BMU*S*S+AH
154             A6=-AA+BMU*S*S+AI
157             A7=S*(AB*GA-AJ)
162             A8=-S*(AB*GA-AK)
165             BA=A1*A6-A2*A5
170             BB=A3*A6-S*A2*A7
173             BC=A4*A6-S*A2*A8
176             BD=S*A1*A7-A3*A5
202             BE=S*A1*A8-A4*A5
205             B1=BB/BA
207             B2=BC/BA
211             B3=BD/BA
212             B4=BE/BA
214             EA=2.*GA*H
216             EB=2.*GB*H
220             EC=(EA+EB)/2.
222             ED=2.*EC
223             E1=EXP(-EA)
226             E2=EXP(-EB)
232             E3=EXP(-EC)
236             E4=EXP(-ED)
242             DL=B2+B3*E4-B4*E2-B1*E1
251             D1=2.*PP/CC/GB/DL
255             D2=AA*AA-S*S*GA*GB
263             D3=B2-B3*E4
266             D4=2.*AA*(GB*(B1*B4-B2*B3)-S*S*GA)*E3
275             D5=(AA*AA+S*S*GA*GB)*(B4*E2-B1*E1)
```

```
306           F=D1*(D2*D3+D4+D5)
313           Z=(F-S)*BESJ0(S*X)*BESJ0(S*Y)
330           RETURN
330           END


              SUBROUTINE LAPINV(GLAM,PHI)
C             THIS PROGRAM EVALUATES THE COEFFICIENTS FOR SERIES
CC            OF JACOBI POLYNOMIALS WHICH REPRESENTS A LAPLACE
C             INVERSION INTEGRAL
              REAL MUL
  5           DIMENSION A(50),GLAM(50),PHI(50),C(4,50)
  5           DIMENSION BK(101),TT(101)
  5           COMMON/2/TI,TF,DT,MN,BK,TT
  5           READ 1,NN,MN,MM
 16         1 FORMAT(3I2)
 16           READ 2,TI,TF,DT
 30         2 FORMAT(3F10.5)
 30           PRINT 99
 34          99 FORMAT(1H1)
 34           CALL SPLICE(GLAM,PHI,MM,C)
 44           PRINT 101
 44         101 FORMAT(/////5X,*    GLAM                PHI     *)
 44           PRINT 102,(GLAM(I),PHI(I),I=1,MM)
 65         102 FORMAT(5X,F10.5,5X,F10.5)
 65           M11=MM-1
 67           PRINT 300
 73         300 FORMAT(////5X,*    C(1,J)          C(2,J)          C(3,J)          C(4
             1,J)  *)
 73           PRINT 103,((C(I,J),I=1,4),J=1,M11)
112         103 FORMAT(5X,F10.5,5X,F10.5,5X,F10.5,5X,F10.5)
112          PRINT 99
116          DO 10 I=1,NN
121          READ 3,BET,DEL
130        3 FORMAT(2F10.5)
130          PRINT 98,BET,DEL
140         98 FORMAT(/////5X,*BETA =*F5.3,* DELTA =*F5.3)
140          DO 11 L=1,MN
143          AL=L
144          S=1./(AL+BET)/DEL
153          CALL SPLINE(GLAM,PHI,MM,C,S,G)
155          F=G*S
161          IF(AL-2.)81,82,83
165         81 A(1)=(1.+BET)*DEL*F
165          GO TO 11
175         82 A(2)=((2.+BET)*DEL*F-A(1))*(3.+BET)
175          GO TO 11
175         83 CONTINUE
177          TOP=1.
201          L1=L-1
202          AL1=L1
203          DO 12 J=1,L1
204          AJ=J
206          TOP=AJ*TOP
210         12 CONTINUE
212          L2=2*L-1
214          BOT=1.
215          DO 13 J=L,L2
216          AJ=J
221          BOT=(AJ+BET)*BOT
223         13 CONTINUE
225          MUL=BOT/TOP
226          SUM=0.0
227          DO 14 N=1,L1
230          AN=N
233          IF(AN-2.)85,86,87
235         85 TOD=1.
237          GO TO 88
237         86 TOD=AL1
237          GO TO 88
241         87 CONTINUE
244          TOD=1.
245          ICH=L1-(N-2)
246          DO 15 J=ICH,L1
             AJ=J
             TOD=AJ*TOD
```

```
250         15  CONTINUE
252         88  CONTINUE
252             BOD=1.
254             JA=L1+N
256             DO 16 J=L,JA
260             AJ=J
261             BOD=BOD*(AJ+BET)
264         16  CONTINUE
266             CO=TOD/BOD
270             SUM=SUM+CO*A(N)
273         14  CONTINUE
275             A(L)=MUL*(DEL*F-SUM)
301         11  CONTINUE
304             CALL JACSER(DEL,A,BET)
306             CALL NAMPLT
307             CALL QIKSET(6.0,0.0,0.0,6.0,0.0,0.0)
313             CALL QIKSAX(3,3)
315             CALL QIKPLT(TT,BK,101)
320             CALL ENDPLT
321         10  CONTINUE
325        999  CONTINUE
325             RETURN
326             END



              SUBROUTINE JACSER(D,C,B)
  6           DIMENSION C(50),SF(50),P(50)
  6           DIMENSION BK(101),TT(101)
  6           COMMON/2/TI,TF,DT,MN,BK,TT
  6           TT(1)=0.0
  7           BK(1)=0.0
 10           LM=1
 11           T=TI
 12          12  T=T+DT
 14              X=2.*EXP(-D*T)-1.
 24              CALL JACOBI(MN,X,B,P)
 26              SF(1)=C(1)*P(1)
 32              DO 10 L=2,MN
 33              L1=L-1
 35              AL=L
 36              SF(L)=SF(L1)+C(L)*P(L)
 43          10  CONTINUE
 45              PRINT 97,T,X
 55          97  FORMAT(/////5X,*  T =*F6.3,*  X =*F10.5)
 55              PRINT 96
 61          96  FORMAT(///5X,* I   C(I)   *,5X,* N   F(T)   *)
 61              DO 11 I=1,6
 65              PRINT 95,I,C(I),I,SF(I)
105          95  FORMAT(5X,I2,F10.2,5X,I2,F10.5)
105          11  CONTINUE
111              LM=LM+1
113              BK(LM)=SF(5)
115              TT(LM)=T
117              IF(T.LE.TF) GO TO 12
121              RETURN
122              END



              SUBROUTINE JACOBI(N,X,B,PB)
          C   THIS PROGRAM CALCULATES JACOBI POLYNOMIALS OF ORDER
          C   K-1 WITH ARG X AND PARAMETER B GT -1
  7           DIMENSION PB(N)
  7           AN=N
 10           IF(AN-2.)1,2,3
 12          1  PB(1)=1.
 14             RETURN
 14          2  PB(1)=1.
 16             PB(2)=X-B*(1.-X)/2.
 31             RETURN
 32          3  BSQ=B*B
 33             BONE=B+1.
 35             PB(1)=1.
 36             PB(2)=X-B*(1.-X)/2.
 51             DO 4 K=3,N
 53             AK=K
```

```
      AK1=AK-1.
      AK2=AK-2.
      K1=K-1
      K2=K-2
      CO1=((2.*AK1)+B)*X
      CO1=((2.*AK2)+B)*CO1
      CO1=((2.*AK2)+BONE)*(CO1-BSQ)
      CO2=2.*AK2*(AK2+B)*((2.*AK1)+B)
      CO=2.*AK1*(AK1+B)*((2.*AK2)+B)
    4 PB(K)=(CO1*PB(K1)-CO2*PB(K2))/CO
      RETURN
      END


      SUBROUTINE SPLINE(X,Y,M,C,XINT,YINT)
      DIMENSION X(50),Y(50),C(4,50)
      IF(XINT-X(1))1,10,11
   10 YINT=Y(1)
      RETURN
   11 CONTINUE
      IF(X(M)-XINT)1,12,13
   12 YINT=Y(M)
      RETURN
   13 CONTINUE
      K=M/2
      N=M
    2 CONTINUE
      IF(X(K)-XINT)3,14,5
   14 YINT=Y(K)
      RETURN
    3 CONTINUE
      IF(XINT-X(K+1))4,15,7
   15 YINT=Y(K+1)
      RETURN
    4 CONTINUE
      YINT=(X(K+1)-XINT)*(C(1,K)*(X(K+1)-XINT)**2+C(3,K))
      YINT=YINT+(XINT-X(K))*(C(2,K)*(XINT-X(K))**2+C(4,K))
      RETURN
    5 CONTINUE
      IF(X(K-1)-XINT)6,16,17
    6 K=K-1
      GO TO 4
   16 YINT=Y(K-1)
      RETURN
   17 N=K
      K=K/2
      GO TO 2
    7 LL=K
      K=(N+K)/2
    8 CONTINUE
      IF(X(K)-XINT)3,14,18
   18 CONTINUE
      IF(X(K-1)-XINT)6,16,19
   19 N=K
      K=(LL+K)/2
      GO TO 8
    1 PRINT 101
  101 FORMAT(* OUT OF RANGE FOR INTERPOLATION  *)
      STOP
      END


      SUBROUTINE SPLICE(X,Y,M,C)
      DIMENSION X(50),Y(50),D(50),P(50),E(50),C(4,50)
      DIMENSION A(50,3),B(50),Z(50)
      MM=M-1
      DO 2 K=1,MM
      D(K)=X(K+1)-X(K)
      P(K)=D(K)/6.
    2 E(K)=(Y(K+1)-Y(K))/D(K)
      DO 3 K=2,MM
    3 B(K)=E(K)-E(K-1)
      A(1,2)=-1.-D(1)/D(2)
      A(1,3)=D(1)/D(2)
      A(2,3)=P(2)-P(1)*A(1,3)
```

```
      A(2,2)=2.*(P(1)+P(2))-P(1)*A(1,2)
      A(2,3)=A(2,3)/A(2,2)
      B(2)=B(2)/A(2,2)
      DO 4 K=3,MM
      A(K,2)=2.*(P(K-1)+P(K))-P(K-1)*A(K-1,3)
      B(K)=B(K)-P(K-1)*B(K-1)
      A(K,3)=P(K)/A(K,2)
    4 B(K)=B(K)/A(K,2)
      Q=D(M-2)/D(M-1)
      A(M,1)=1.+Q+A(M-2,3)
      A(M,2)=-Q-A(M,1)*A(M-1,3)
      B(M)=B(M-2)-A(M,1)*B(M-1)
      Z(M)=B(M)/A(M,2)
      MN=M-2
      DO 6 I=1,MN
      K=M-I
    6 Z(K)=B(K)-A(K,3)*Z(K+1)
      Z(1)=-A(1,2)*Z(2)-A(1,3)*Z(3)
      DO 7 K=1,MM
      Q=1./(6.*D(K))
      C(1,K)=Z(K)*Q
      C(2,K)=Z(K+1)*Q
      C(3,K)=Y(K)/D(K)-Z(K)*P(K)
    7 C(4,K)=Y(K+1)/D(K)-Z(K+1)*P(K)
      RETURN
      END
```

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